適航指令發布單
Airworthiness Directive Issuance Form

民航局AD編號 AD number	CAA-2026-11-016	發布日期 Date issued	2026/11/23
適用之航空產品 Applied to (models, serial numbers or part numbers, as applicable)	BR700-725A1-12 engines, all serial numbers. These engines are known to be installed on, but not limited to, Gulfstream GVI (commercial designation G650) aeroplanes.		
主旨摘要 Subject	Time Limits / Maintenance Checks - Time Limits Manual / Maintenance Programme - Amendment		

民航局
CAA

○ 本國產品
Native product

○ 其他個案
Other

設計國民航主管機構
Original Authority

- ○ FAA
- ◉ EASA
- ○ Brazil
- ○ Transport Canada Civil Aviation
- ○ DGAC
- ○ Germany LBA
- ○ CAA-NL
- ○ UK CAA
- ○ Japan CAB
- ○ CAA of Israel
- ○ Other____

設計國AD編號 Original AD number	2024-0136

1. 直接採用原AD之內容? (Is the original AD directly adopted?)
   ◉ 是(Yes)　○ 否(No)_
   a. 生效日期另訂為(Re-specify the effective date as):
   b. 執行時限另訂為(Re-specify the compliance time or period as):
2. 使用人是否需要將AD執行結果向民航局提出報告?
   (Do users need to report the status of compliance to the CAA?)
   ○ 需要(Yes)　◉ 不需要(No)

備註 Note	ATA 72. Ref. Publications: RRD BR700-725A1-12 TLM T-725-7BR Revision 27 dated 15 April 2024.

註：
1. AD內容後附。
2. 航空器產品使用人得向民航局提出豁免、替代符合方法、執行時限之展延之申請。
3. 如有任何問題，請聯絡交通部民用航空局初始適航科。Tel：(02)2349-6330 / 6332, Fax：(02)2545-8464, adcaa@mail.caa.gov.tw

Note：
1. The AD text is enclosed.
2. Exemption, an alternative method of compliance or adjustment of the compliance time may be proposed to the CAA for approval.
3. For further information, please contact Civil Aviation Administration on Tel：(02)2349-6330 / 6332, Fax：(02)2545-8464, adcaa@mail.caa.gov.tw

Airworthiness Directive

AD No.: **2024-0136**

Issued: **11 July 2024**

Note: This Airworthiness Directive (AD) is issued by EASA, acting in accordance with Regulation (EU) 2018/1139 on behalf of the European Union, its Member States and of the European third countries that participate in the activities of EASA under Article 129 of that Regulation.

This AD is issued in accordance with Regulation (EU) 748/2012, Part 21.A.3B. In accordance with Regulation (EU) 1321/2014 Annex I Part M.A.301, or Annex Vb Part ML.A.301, as applicable, the continuing airworthiness of an aircraft shall be ensured by accomplishing any applicable ADs. Consequently, no person may operate an aircraft to which an AD applies, except in accordance with the requirements of that AD, unless otherwise specified by the Agency [Regulation (EU) 1321/2014 Annex I Part M.A.303, or Annex Vb Part ML.A.303, as applicable] or agreed with the Authority of the State of Registry [Regulation (EU) 2018/1139, Article 71 exemption].

Design Approval Holder's Name:
ROLLS-ROYCE DEUTSCHLAND Ltd & Co KG

Type/Model designation(s):
BR700-725A1-12 engines

Effective Date: 25 July 2024
TCDS Number(s): EASA.E.018
Foreign AD: Not applicable
Supersedure: None

ATA 05 – Time Limits / Maintenance Checks – Time Limits Manual / Maintenance Programme – Amendment

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Manufacturer(s):
Rolls-Royce Deutschland Ltd & Co KG (RRD)

Applicability:
BR700-725A1-12 engines, all serial numbers.

These engines are known to be installed on, but not limited to, Gulfstream GVI (commercial designation G650) aeroplanes.

Definitions:
For the purpose of this AD, the following definitions apply:

The TLM: RRD BR700-725A1-12 Time Limits Manual (TLM) T-725-7BR at Revision 27.

The AMP: The Aircraft Maintenance Programme (AMP) contains the tasks on the basis of which the scheduled maintenance is conducted to ensure the continuing airworthiness of each operated aircraft. For engines installed on aircraft operated under EU regulations, the operator or the owner ensures compliance with the AMP as stipulated in Commission Regulation (EU) 1321/2014.

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New and/or more restrictive tasks and limitations: This includes all tasks and limitations that are new and all tasks for which a threshold or interval was reduced, which were introduced into the TLM, as defined in this AD, since the previous TLM Revision that is currently incorporated in the AMP.

Reason:

The Airworthiness Limitations Section instructions for BR700-725A1-12 engines, which are approved by EASA, are defined and published in TLM T-725-7BR. These instructions have been identified as mandatory for continued airworthiness.

Failure to accomplish these instructions could result in an unsafe condition.

For the reason described above, this AD requires accomplishment of the actions and limitations specified in the TLM.

Required Action(s) and Compliance Time(s):

Required as indicated by this AD, unless the action(s) required by this AD have been already accomplished:

Maintenance Tasks and Replacement of Life Limited Parts:

(1) From the effective date of this AD, accomplish the following actions, as specified in the TLM, as applicable to engine model and depending on engine configuration:

(1.1) Replace each component before exceeding the applicable life limit, and

(1.2) Within the thresholds and intervals, accomplish all applicable maintenance tasks.

Corrective Action(s):

(2) In case of finding discrepancies (as defined in the TLM) during accomplishment of any action as required by paragraph (1) of this AD, within the compliance time specified in the TLM, accomplish the applicable corrective action(s) in accordance with approved RRD instructions for continued airworthiness. If no compliance time is identified in the TLM, accomplish the applicable corrective action(s) before next flight. If a detected discrepancy is not identified in the TLM, before next flight, contact RRD for approved instructions and accomplish those instructions accordingly.

AMP Revision:

(3) Within 12 months after the effective date of this AD, revise the approved AMP by incorporating the limitations, tasks and associated thresholds and intervals described in the TLM, as applicable to engine model and depending on engine configuration.

Credit:

(4) If, before the effective date of this AD, the AMP has been revised to incorporate the maintenance tasks and life limitations as specified in the previous TLM revision, that action ensures the continued accomplishment of those tasks and limitations.

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Consequently, for an engine to which that AMP applies, it is acceptable to accomplish the new and/or more restrictive tasks and limitations as specified in the TLM, as applicable to engine model and depending on engine configuration, within the compliance times as specified in the TLM, to comply with paragraph (1) of this AD.

For that AMP, it is acceptable to incorporate the new and/or more restrictive tasks and limitations as specified in the TLM, as applicable to engine model and depending on engine configuration, into the AMP to comply with paragraph (3) of this AD.

Recording AD Compliance:

(5) When the AMP of an aircraft has been revised as required by paragraph (3) or (4) of this AD, as applicable, that action ensures continued accomplishment of the tasks and limitations as required by paragraphs (1) and (2) of this AD for that engine. Consequently, after revising the AMP, as required by paragraph (3) or (4) of this AD, as applicable, it is not necessary that accomplishment of individual action is recorded for demonstration of AD compliance on a continued basis.

Ref. Publications:

RRD BR700-725A1-12 TLM T-725-7BR Revision 27 dated 15 April 2024.

The use of later approved revisions of the above-mentioned document is acceptable for compliance with the requirements of this AD.

Remarks:

1. If requested and appropriately substantiated, EASA can approve Alternative Methods of Compliance for this AD.

2. Based on the required actions and the compliance time, EASA have decided to issue a Final AD with Request for Comments, postponing the public consultation process until after publication. All interested persons may send their comments, referencing the AD Number, to the E-mail address specified in below Remark 3, prior to 25 August 2024. Only if any comment is received during the consultation period, a Comment Response Document will be published in the EASA Safety Publications Tool, in a compressed ('zipped') file, attached to the record for this AD.

3. Enquiries regarding this AD should be referred to the EASA Safety Information Section, Certification Directorate. E-mail: ADs@easa.europa.eu.

4. Information about any failures, malfunctions, defects or other occurrences, which may be similar to the unsafe condition addressed by this AD, and which may occur, or have occurred on a product, part or appliance not affected by this AD, can be reported to the EU aviation safety reporting system. This may include reporting on the same or similar components, other than those covered by the design to which this AD applies, if the same unsafe condition can exist or may develop on an aircraft with those components installed. Such components may be installed under an FAA Parts Manufacturer Approval (PMA), Supplemental Type Certificate (STC) or other modification.

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5. For any question concerning the technical content of the requirements in this AD, please contact:
   Rolls-Royce Deutschland Ltd & Co KG,
   Eschenweg 11, Dahlewitz 15827 Blankenfelde-Mahlow, Germany
   Telephone: +49 (0) 33 7086 1200 Fax: +49 (0) 33 7086 1212
   E-mail: RRDTechnicalHelpdesk@Rolls-Royce.com.

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