適航指令發布單
Airworthiness Directive Issuance Form

民航局AD編號 AD number	CAA-2026-11-003	發布日期 Date issued	2026/11/10
適用之航空產品 Applied to (models, serial numbers or part numbers, as applicable)	MBB-BK117 D-2 (all variants) and D-3 helicopters, serial numbers 20001 to 20360 inclusive.		
主旨摘要 Subject	Landing Gear - Emergency Flotation System Adhesive Label - Removal		

民航局
CAA

○ 本國產品
Native product

○ 其他個案
Other

設計國民航主管機構
Original Authority

- ○ FAA
- ◉ EASA
- ○ Brazil
- ○ Transport Canada Civil Aviation
- ○ DGAC
- ○ Germany LBA
- ○ CAA-NL
- ○ UK CAA
- ○ Japan CAB
- ○ CAA of Israel
- ○ Other____

設計國AD編號 Original AD number	2026-0201

1. 直接採用原AD之內容? (Is the original AD directly adopted?)
   ◉ 是(Yes) ○ 否(No)_
   a. 生效日期另訂為(Re-specify the effective date as):
   b. 執行時限另訂為(Re-specify the compliance time or period as):
2. 使用人是否需要將AD執行結果向民航局提出報告?
   (Do users need to report the status of compliance to the CAA?)
   ○ 需要(Yes) ◉ 不需要(No)

備註 Note	ATA 32. Ref. Publications: AH ASB MBB-BK117-25-67-0006 original issue (Issue 001) dated 08 September 2026.

註： 1. AD內容後附。
2. 航空器產品使用人得向民航局提出豁免、替代符合方法、執行時限之展延之申請。
3. 如有任何問題，請聯絡交通部民用航空局初始適航科。Tel：(02)2349-6330 / 6332, Fax：(02)2545-8464, adcaa@mail.caa.gov.tw

Note：1. The AD text is enclosed.
2. Exemption, an alternative method of compliance or adjustment of the compliance time may be proposed to the CAA for approval.
3. For further information, please contact Civil Aviation Administration on Tel：(02)2349-6330 / 6332, Fax：(02)2545-8464, adcaa@mail.caa.gov.tw

European Union Aviation Safety Agency

Airworthiness Directive

AD No.: 2026-0201

Issued: 26 October 2026

Note: This Airworthiness Directive (AD) is issued by EASA, acting in accordance with Regulation (EU) 2018/1139 on behalf of the European Union, its Member States and of the European third countries that participate in the activities of EASA under Article 129 of that Regulation.

This AD is issued in accordance with Regulation (EU) 748/2012, Part 21.A.3B. In accordance with Regulation (EU) 1321/2014 Annex I Part M.A.301, or Annex Vb Part ML.A.301, as applicable, the continuing airworthiness of an aircraft shall be ensured by accomplishing any applicable ADs. Consequently, no person may operate an aircraft to which an AD applies, except in accordance with the requirements of that AD, unless otherwise specified by the Agency [Regulation (EU) 1321/2014 Annex I Part M.A.303, or Annex Vb Part ML.A.303, as applicable] or agreed with the Authority of the State of Registry [Regulation (EU) 2018/1139, Article 71 exemption].

Design Approval Holder's Name:	Type/Model designation(s):
AIRBUS HELICOPTERS DEUTSCHLAND GmbH	MBB-BK117 helicopters

Effective Date:	09 November 2026
TCDS Number(s):	EASA.R.010
Foreign AD:	Not applicable
Supersedure:	None

ATA 32 – Landing Gear – Emergency Flotation System Adhesive Label – Removal

---

Manufacturer(s):

Airbus Helicopters (AH); Airbus Helicopters Deutschland GmbH; Kawasaki Heavy Industries Ltd.; and Airbus Helicopters Inc.

Applicability:

MBB-BK117 D-2 (all variants) and D-3 helicopters, serial numbers 20001 to 20360 inclusive.

Definitions:

For the purpose of this AD, the following definitions apply:

The ASB: AH Alert Service Bulletin (ASB) MBB-BK117-25-67-0006.

Affected label: Emergency Flotation System (EFS) adhesive label Part Number 214794-0.

Groups: Group 1 helicopters are those that have an affected label installed. Groups 2 helicopters are those that do not have an affected label installed.

Reason:

An investigation has revealed that a non-certified EFS adhesive label, displaying incorrect EFS gas cylinder pressure limits, was listed in the MBB-BK117 D-2 Illustrated Parts Catalogue and therefore may have been installed on the structure of the helicopter.

An agency of the European Union

TE.CAP.00110-012 © European Union Aviation Safety Agency. All rights reserved. ISO9001 Certified.
Proprietary document. Copies are not controlled. Confirm revision status through the EASA-Internet/Intranet.

This condition, if not corrected, could lead to an incomplete inflation of the floats during ditching, potentially preventing timely evacuation from the helicopter and consequent injury to occupants.

To address this potential unsafe condition, AH issued the ASB, to provide instructions for removal of the affected label.

For the reason described above, this AD requires removal of those affected labels.

Required Action(s) and Compliance Time(s):

Required as indicated by this AD, unless the action(s) required by this AD have been already accomplished:

Removal:

(1) For Group 1 helicopters: Within 18 months after the effective date of this AD, remove the affected label in accordance with the instructions of the ASB.

Part(s) Installation:

(2) For Group 1 and Group 2 helicopters: From the effective date of this AD, do not install an affected label on any helicopter.

Ref. Publications:

AH ASB MBB-BK117-25-67-0006 original issue (Issue 001) dated 08 September 2026.

The use of later approved revisions of the above-mentioned document is acceptable for compliance with the requirements of this AD.

Remarks:

1. If requested and appropriately substantiated, EASA can approve Alternative Methods of Compliance for this AD.

2. The original issue of this AD was posted on 22 September 2026 as PAD 26-111 for consultation until 20 October 2026. No comments were received during the consultation period.

3. Enquiries regarding this AD should be referred to the EASA Safety Information Section, Certification Directorate. E-mail: ADs@easa.europa.eu.

4. Information about any failures, malfunctions, defects or other occurrences, which may be similar to the unsafe condition addressed by this AD, and which may occur, or have occurred on a product, part or appliance not affected by this AD, can be reported to the EU aviation safety reporting system. This may include reporting on the same or similar components, other than those covered by the design to which this AD applies, if the same unsafe condition can exist or may develop on an aircraft with those components installed. Such components may be installed under an FAA Parts Manufacturer Approval (PMA), Supplemental Type Certificate (STC) or other modification.

An agency of the European Union

TE.CAP.00110-012 © European Union Aviation Safety Agency. All rights reserved. ISO9001 Certified.
Proprietary document. Copies are not controlled. Confirm revision status through the EASA-Internet/Intranet.

5. For any question concerning the technical content of the requirements in this AD, please contact: Airbus Helicopters (Technical Support) at:
Web portal: https://airbusworld.helicopters.airbus.com / Technical Requests Management, or
E-mail: TechnicalSupport.Helicopters@airbus.com, or Telephone: +33 (0)4 42 859 789.

An agency of the European Union

TE.CAP.00110-012 © European Union Aviation Safety Agency. All rights reserved. ISO9001 Certified.
Proprietary document. Copies are not controlled. Confirm revision status through the EASA-Internet/Intranet.