適航指令發布單
Airworthiness Directive Issuance Form

民航局AD編號 AD number	CAA-2026-11-002	發布日期 Date issued	2026/11/03
適用之航空產品 Applied to (models, serial numbers or part numbers, as applicable)	Airbus A330-841 and A330-941 aeroplanes, manufacturer serial numbers (MSN) 1819, 1836, 1844, 1850, 1861, 1870, 1876, 1884, 1887, 1890, 1892, 1895, 1897, 1900, 1901, 1903, 1904, 1906, 1908, 1909, 1910, 1913, 1914, 1915, 1918, 1920, 1921, 1922, 1923, 1924, 1925, 1926, 1927, 1929, 1931, 1933, 1934, 1935, 1937, 1938, 1939, 1941, 1943, 1946, 1947, 1948, 1949, 1952, 1953, 1954, 1956, 1957, 1958, 1959, 1961, 1962, 1963, 1964, 1966, 1969, 1971, 1972, 1977 and 1979.		
主旨摘要 Subject	Landing Gear - Tachometer Wiring Harness - Modification		

民航局
CAA

○ 本國產品
Native product

○ 其他個案
Other

設計國民航主管機構
Original Authority

○ FAA	○ Germany LBA
◉ EASA	○ CAA-NL
○ Brazil	○ UK CAA
○ Transport Canada Civil Aviation	○ Japan CAB
○ DGAC	○ CAA of Israel
	○ Other____

設計國AD編號 Original AD number	2026-0200

1. 直接採用原AD之內容? (Is the original AD directly adopted?)
   ◉ 是(Yes) ○ 否(No)_
   a. 生效日期另訂為(Re-specify the effective date as):
   b. 執行時限另訂為(Re-specify the compliance time or period as):

2. 使用人是否需要將AD執行結果向民航局提出報告?
   (Do users need to report the status of compliance to the CAA?)
   ○ 需要(Yes) ◉ 不需要(No)

備註 Note	ATA 32. Ref. Publications: Airbus SB A330-32-3306 original issue dated 24 July 2026.

註： 1. AD內容後附。
2. 航空器產品使用人得向民航局提出豁免、替代符合方法、執行時限之展延之申請。
3. 如有任何問題，請聯絡交通部民用航空局初始適航科。Tel：(02)2349-6330 / 6332, Fax：(02)2545-8464, adcaa@mail.caa.gov.tw

Note：1. The AD text is enclosed.
2. Exemption, an alternative method of compliance or adjustment of the compliance time may be proposed to the CAA for approval.
3. For further information, please contact Civil Aviation Administration on Tel：(02)2349-6330 / 6332, Fax：(02)2545-8464, adcaa@mail.caa.gov.tw

Airworthiness Directive

AD No.: 2026-0200

Issued: 20 October 2026

Note: This Airworthiness Directive (AD) is issued by EASA, acting in accordance with Regulation (EU) 2018/1139 on behalf of the European Union, its Member States and of the European third countries that participate in the activities of EASA under Article 129 of that Regulation.

This AD is issued in accordance with Regulation (EU) 748/2012, Part 21.A.3B. In accordance with Regulation (EU) 1321/2014 Annex I Part M.A.301, or Annex Vb Part ML.A.301, as applicable, the continuing airworthiness of an aircraft shall be ensured by accomplishing any applicable ADs. Consequently, no person may operate an aircraft to which an AD applies, except in accordance with the requirements of that AD, unless otherwise specified by the Agency [Regulation (EU) 1321/2014 Annex I Part M.A.303, or Annex Vb Part ML.A.303, as applicable] or agreed with the Authority of the State of Registry [Regulation (EU) 2018/1139, Article 71 exemption].

Design Approval Holder's Name:

AIRBUS S.A.S.

Type/Model designation(s):

A330-841 and A330-941 aeroplanes

Effective Date: 03 November 2026

TCDS Number(s): EASA.A.004

Foreign AD: Not applicable

Supersedure: None

ATA 32 – Landing Gear – Tachometer Wiring Harness – Modification

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Manufacturer(s):

Airbus

Applicability:

Airbus A330-841 and A330-941 aeroplanes, manufacturer serial numbers (MSN) 1819, 1836, 1844, 1850, 1861, 1870, 1876, 1884, 1887, 1890, 1892, 1895, 1897, 1900, 1901, 1903, 1904, 1906, 1908, 1909, 1910, 1913, 1914, 1915, 1918, 1920, 1921, 1922, 1923, 1924, 1925, 1926, 1927, 1929, 1931, 1933, 1934, 1935, 1937, 1938, 1939, 1941, 1943, 1946, 1947, 1948, 1949, 1952, 1953, 1954, 1956, 1957, 1958, 1959, 1961, 1962, 1963, 1964, 1966, 1969, 1971, 1972, 1977 and 1979.

Definitions:

For the purpose of this AD, the following definitions apply:

None.

Reason:

A design review determined that the duplication of the routing of the tachometer wiring harnesses, introduced to address the threat of UERF (Uncontained Engine Rotor Failure), created a vulnerability, while being alongside the blue hydraulic pipe. Following a tire tread separation event, some debris could damage both tachometers' wiring and the blue hydraulic pipe, possibly leading to loss of Normal and Alternate braking function.

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This condition, if not corrected, could lead to inability to stop the aeroplane.

To address this potential unsafe condition, Airbus issued instructions to modify the routing of the tachometer wiring harnesses.

For the reason described above, this AD requires modification of the routing of the of the 2M tachometer wiring harnesses.

Required Action(s) and Compliance Time(s):

Required as indicated, by this AD, unless the action(s) required by this AD have been already accomplished:

Modification:

(1) Within 61 months after the effective date of this AD, modify the aeroplane in accordance with the instructions of Airbus SB A330-32-3306.

(2) [RESERVED].

Ref. Publications:

Airbus SB A330-32-3306 original issue dated 24 July 2026.

The use of later approved revisions of the above-mentioned document is acceptable for compliance with the requirements of this AD.

Remarks:

1. If requested and appropriately substantiated, EASA can approve Alternative Methods of Compliance for this AD.

2. This AD was posted on 07 September 2026 as PAD 26-101 for consultation until 05 October 2026. The Comment Response Document can be found in the EASA Safety Publications Tool, in the compressed (zipped) file attached to the record for this AD.

3. Enquiries regarding this AD should be referred to the EASA Safety Information Section, Certification Directorate. E-mail: ADs@easa.europa.eu.

4. Information about any failures, malfunctions, defects or other occurrences, which may be similar to the unsafe condition addressed by this AD, and which may occur, or have occurred on a product, part or appliance not affected by this AD, can be reported to the EU aviation safety reporting system. This may include reporting on the same or similar components, other than those covered by the design to which this AD applies, if the same unsafe condition can exist or may develop on an aircraft with those components installed. Such components may be installed under an FAA Parts Manufacturer Approval (PMA), Supplemental Type Certificate (STC) or other modification.

5. For any question concerning the technical content of the requirements in this AD, please contact: AIRBUS – 1IAL (Airworthiness Office), E-mail: airworthiness.A330-A340@airbus.com.

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