



適航指令發布單

Airworthiness Directive Issuance Form

民航局AD編號 AD number	CAA-2026-05-009緊急	發布日期 Date issued	2026/05/30
適用之航空產品 Applied to (models, serial numbers or part numbers, as applicable)	Pratt & Whitney Canada (P&WC) Models PW210A, PW210A1, and PW210S engines, all serial numbers.		
主旨摘要 Subject	Engine - Exhaust - Turbine Exhaust Frame (TEF) Cracks		
民航局 CAA	設計國民航主管機構 Original Authority		
<input type="radio"/> 本國產品 Native product	<input type="radio"/> FAA	<input type="radio"/> Germany LBA	
<input type="radio"/> 其他個案 Other	<input type="radio"/> EASA	<input type="radio"/> CAA-NL	
	<input type="radio"/> Brazil	<input type="radio"/> UK CAA	
	<input checked="" type="radio"/> Transport Canada Civil Aviation	<input type="radio"/> Japan CAB	
	<input type="radio"/> DGAC	<input type="radio"/> CAA of Israel	
		<input type="radio"/> Other_____	
	設計國AD編號 Original AD number	CF-2026-23	
	1. 直接採用原AD之內容? (Is the original AD directly adopted?) <input checked="" type="radio"/> 是(Yes) <input type="radio"/> 否(No)_ a. 生效日期另訂為(Re-specify the effective date as): b. 執行時限另訂為(Re-specify the compliance time or period as):		
	2. 使用人是否需要將AD執行結果向民航局提出報告? (Do users need to report the status of compliance to the CAA?) <input type="radio"/> 需要(Yes) <input checked="" type="radio"/> 不需要(No)		
備註 Note	ATA72. Ref.P&WC ASB: PW210-72-A57186, Revision 2, dated 26 May 2026, or later revisions approved by the Chief, Continuing Airworthiness, Transport Canada.		

註： 1. AD內容後附。

2. 航空器產品使用人得向民航局提出豁免、替代符合方法、執行時限之展延之申請。

3. 如有任何問題，請聯絡交通部民用航空局初始適航科。Tel：(02)2349-6330 / 6332, Fax：(02)2545-8464,

adcaa@mail.caa.gov.tw

Note： 1. The AD text is enclosed.

2. Exemption, an alternative method of compliance or adjustment of the compliance time may be proposed to the CAA for approval.

3. For further information, please contact Civil Aviation Administration on Tel：(02)2349-6330 / 6332,

Fax：(02)2545-8464, adcaa@mail.caa.gov.tw



EMERGENCY AIRWORTHINESS DIRECTIVE

This Airworthiness Directive (AD) is issued pursuant to Canadian Aviation Regulation (CAR) 521.427. No person shall conduct a take-off or permit a take-off to be conducted in an aircraft that is in their legal custody and control, unless the requirements of CAR 605.84 pertaining to ADs are met. Standard 625 - Aircraft Equipment and Maintenance Standards Appendix H provides information concerning alternative means of compliance (AMOC) with ADs.

Number:

CF-2026-23

Effective Date:

1 June 2026

ATA:

72

Type Certificate:

E-36

Subject:

Engine – Exhaust – Turbine Exhaust Frame (TEF) Cracks

Applicability:

Pratt & Whitney Canada (P&WC) Models PW210A, PW210A1, and PW210S engines, all serial numbers.

Compliance:

As indicated below, unless already accomplished.

Background:

Operators reported circumferential cracks at the TEF external surface. Investigation revealed that the cracks originated from a wall of the TEF which was found under the minimum required thickness. Subsequent crack initiation and growth were associated with thermal stress linked to engine starts. Failure of the TEF could lead to a loss of axial containment and subsequent release of parts and damage to the helicopter.

To mitigate this risk, P&WC published Alert Service Bulletin (ASB) PW210-72-A57186 to require an initial and repetitive general visual inspection (GVI) of the TEF. This AD mandates the corrective actions of P&WC PW210-72-A57186.

This AD is considered an interim action, and further AD action may follow.

Corrective Actions:

For the purpose of this AD, the following definition applies:

P&WC ASB: PW210-72-A57186, Revision 2, dated 26 May 2026, or later revisions approved by the Chief, Continuing Airworthiness, Transport Canada.

Part I – Initial GVI

- A. For engines that have accumulated 7000 cycles since new or more as of the effective date of this AD, within 5 hours air time or 20 engine starts, whichever occurs first, perform a GVI of the TEF, in accordance with Paragraph A. of the Accomplishment Instructions of the P&WC ASB.
- B. For engines that have accumulated 2000 hours air time or more and less than 7000 cycles since new as of the effective date of this AD, within 50 hours air time or 200 engine starts, whichever occurs first, perform a GVI of the TEF, in accordance with Paragraph A. of the Accomplishment Instructions of the P&WC ASB.
- C. If the combined length of crack(s) exceeded 9.5 inches, before further flight, replace the part. Otherwise, proceed to Part II of this AD.

Part II – Repetitive Inspection and Part Replacement

Following the completion of Part I of this AD, repeat the GVI of the TEF, in accordance with Paragraph A. of the Accomplishment Instructions of the P&WC ASB or replace the part, as applicable. After each GVI:

- A. If no cracks were found, repeat the GVI within 200 engine starts.
- B. If the combined length of crack(s) was less than 9.5 inches, repeat the GVI within 20 engine starts.
- C. If the combined length of crack(s) exceeded 9.5 inches, before further flight, replace the part.

Authorization:

For the Minister of Transport,

ORIGINAL SIGNED BY

Jenny Young
Chief, Continuing Airworthiness
Issued on 29 May 2026

Contact:

Isabelle Hallé, Continuing Airworthiness, Ottawa, telephone 1-888-663-3639, or e-mail TC.AirworthinessDirectives-Consignesdenavigabilite.TC@tc.gc.ca or any Transport Canada Centre.