



# 適航指令發布單

## Airworthiness Directive Issuance Form

民航局AD編號 AD number	CAA-2025-02-011	發布日期 Date issued	2025/02/14
適用之航空產品 Applied to (models, serial numbers or part numbers, as applicable)	Airbus A350-941 and A350-1041 aeroplanes, all manufacturer serial numbers (MSN), except those on which Airbus modification (mod) 116830 has been embodied in production.		
主旨摘要 Subject	Hydraulic Power - Engine Driven Pump - Software Update / Modification		
民航局 CAA	設計國民航主管機構 Original Authority		
<input type="radio"/> 本國產品 Native product	<input type="radio"/> FAA	<input type="radio"/> Germany LBA	<input type="radio"/> CAA-NL
<input type="radio"/> 其他個案 Other	<input checked="" type="radio"/> EASA	<input type="radio"/> UK CAA	<input type="radio"/> Japan CAB
	<input type="radio"/> Brazil	<input type="radio"/> Transport Canada Civil Aviation	<input type="radio"/> CAA of Israel
	<input type="radio"/> DGAC	<input type="radio"/> Other _____	
	設計國AD編號 Original AD number	2025-0035	
	1. 直接採用原AD之內容? (Is the original AD directly adopted?) <input checked="" type="radio"/> 是(Yes) <input type="radio"/> 否(No)_ a. 生效日期另訂為(Re-specify the effective date as): b. 執行時限另訂為(Re-specify the compliance time or period as):		
	2. 使用人是否需要將AD執行結果向民航局提出報告? (Do users need to report the status of compliance to the CAA?) <input type="radio"/> 需要(Yes) <input checked="" type="radio"/> 不需要(No)		
備註 Note	ATA 29. Ref. Publications: Airbus SB A350-29-P051 original issue dated 28 October 2024 or revision 01 dated December 2024. and Airbus SB A350-29-P052 original issue dated 28 October 2024. and Airbus SB A350-42-P020 original issue dated 10 August 2023.		

註： 1. AD內容後附。  
2. 航空器產品使用人得向民航局提出豁免、替代符合方法、執行時限之展延之申請。  
3. 如有任何問題，請聯絡交通部民用航空局初始適航科。Tel：(02)2349-6330 / 6332, Fax：(02)2545-8464,  
[adcaa@mail.caa.gov.tw](mailto:adcaa@mail.caa.gov.tw)

Note： 1. The AD text is enclosed.  
2. Exemption, an alternative method of compliance or adjustment of the compliance time may be proposed to  
the CAA for approval.  
3. For further information, please contact Civil Aviation Administration on Tel：(02)2349-6330 / 6332,  
Fax：(02)2545-8464, [adcaa@mail.caa.gov.tw](mailto:adcaa@mail.caa.gov.tw)



## Airworthiness Directive

**AD No.:** 2025-0035

**Issued:** 12 February 2025

Note: This Airworthiness Directive (AD) is issued by EASA, acting in accordance with Regulation (EU) 2018/1139 on behalf of the European Union, its Member States and of the European third countries that participate in the activities of EASA under Article 129 of that Regulation.

This AD is issued in accordance with Regulation (EU) 748/2012, Part 21.A.3B. In accordance with Regulation (EU) 1321/2014 Annex I Part M.A.301, or Annex Vb Part ML.A.301, as applicable, the continuing airworthiness of an aircraft shall be ensured by accomplishing any applicable ADs. Consequently, no person may operate an aircraft to which an AD applies, except in accordance with the requirements of that AD, unless otherwise specified by the Agency [Regulation (EU) 1321/2014 Annex I Part M.A.303, or Annex Vb Part ML.A.303, as applicable] or agreed with the Authority of the State of Registry [Regulation (EU) 2018/1139, Article 71 exemption].

**Design Approval Holder's Name:**

AIRBUS S.A.S.

**Type/Model designation(s):**

A350 aeroplanes

**Effective Date:** 26 February 2025

**TCDS Number(s):** EASA.A.151

**Foreign AD:** Not applicable

**Supersedure:** None

### ATA 29 – Hydraulic Power – Engine Driven Pump – Software Update / Modification

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**Manufacturer(s):**

Airbus

**Applicability:**

Airbus A350-941 and A350-1041 aeroplanes, all manufacturer serial numbers (MSN), except those on which Airbus modification (mod) 116830 has been embodied in production.

**Definitions:**

For the purpose of this AD, the following definitions apply:

**The SB1:** Airbus Service Bulletin (SB) A350-29-P051 revision 01.

**The SB2:** Airbus SB A350-29-P052.

**Groups:** Group 1 aeroplanes are those that have Hydraulic Monitoring and Control Application (HMCA) software (SW) S4.2 standard installed.

Group 2 aeroplanes are those that have HMCA SW S5.0 standard installed.

**Reason:**

Occurrences of failure of hydraulic engine driven pumps (EDP) of the standard EDP-06 were reported. The consequent high friction and subsequent damage to the EDP could lead to



overheating of the EDP Case Drain (CD). As per design, in case of overheat, a temperature monitoring and isolation of the hydraulic fluid CD cooling line, located in the fuel tanks, is in place.

This condition, if not corrected, leads to a non-compliance to the quantitative safety requirements for the “uncontrolled EDP CD overheat” failure condition, due to the repetitive nature of those overheat events.

To address this potential unsafe condition, Airbus developed Hydraulic Monitoring and Control Application (HMCA) SW S6.0 enhancing the robustness of the EPD CD overheat protection, which is embodied in production through Airbus mod 116830, and published Airbus SB1 and SB2 for retrofit installation.

For the reasons described above, this AD requires modification of the HMCA.

This AD is considered to be an interim action and further AD action may follow.

#### Required Action(s) and Compliance Time(s):

Required as indicated by this AD, unless the action(s) required by this AD have been already accomplished:

#### Modification:

- (1) Within the compliance time specified in Table 1 of this AD, as applicable, modify the HMCA in accordance with the instructions of the applicable SB (see Note 1 of this AD).

Table 1 – Modification

Group	Applicable SB	Compliance Time (after the effective date of this AD)
1	SB1	Within 6 months
2	SB2	Within 12 months

Note 1: After accomplishment of the modification as required by paragraph (1) of this AD, an aeroplane remains in compliance with the requirement of paragraph (3) of EASA AD 2018-0178, as applicable.

#### Concurrent Requirements / Additional Modification:

- (2) Prior to or concurrent with modification of an aeroplane as required by paragraph (1) of this AD, modify that aeroplane in accordance with the instructions of Airbus SB A350-42-P020.

#### Credit:

- (3) Modification of an aeroplane, accomplished before the effective date of this AD in accordance with the instructions of SB A350-29-P051 at original issue and RDAF 81504363/023/2024#C is acceptable to comply with the requirement of paragraph (1) of this AD for that aeroplane.



**Ref. Publications:**

Airbus SB A350-29-P051 original issue dated 28 October 2024 or revision 01 dated December 2024.

Airbus SB A350-29-P052 original issue dated 28 October 2024.

Airbus SB A350-42-P020 original issue dated 10 August 2023.

The use of later approved revisions of the above-mentioned documents is acceptable for compliance with the requirements of this AD.

**Remarks:**

1. If requested and appropriately substantiated, EASA can approve Alternative Methods of Compliance for this AD.
2. This AD was posted on 04 December 2024 as PAD 24-146 for consultation until 03 January 2025, and republished on 22 January 2025 as PAD 24-146R1 for additional consultation until 05 February 2025. The Comment Response Documents can be found in the [EASA Safety Publications Tool](#), in the compressed (zipped) file attached to the record for this AD.
3. Enquiries regarding this AD should be referred to the EASA Safety Information Section, Certification Directorate. E-mail: [ADs@easa.europa.eu](mailto:ADs@easa.europa.eu).
4. Information about any failures, malfunctions, defects or other occurrences, which may be similar to the unsafe condition addressed by this AD, and which may occur, or have occurred on a product, part or appliance not affected by this AD, can be reported to the [EU aviation safety reporting system](#). This may include reporting on the same or similar components, other than those covered by the design to which this AD applies, if the same unsafe condition can exist or may develop on an aircraft with those components installed. Such components may be installed under an FAA Parts Manufacturer Approval (PMA), Supplemental Type Certificate (STC) or other modification.
5. For any question concerning the technical content of the requirements in this AD, please contact: AIRBUS S.A.S. A350 XWB (1IAK), E-mail: [continued-airworthiness.a350@airbus.com](mailto:continued-airworthiness.a350@airbus.com).

