



適航指令發布單

Airworthiness Directive Issuance Form

民航局AD編號 AD number	CAA-2025-01-003	發布日期 Date issued	2025/01/09
適用之航空產品 Applied to (models, serial numbers or part numbers, as applicable)	ATR 42-400 and ATR 42-500 aeroplanes, all manufacturer serial numbers (MSN), except those on which ATR modification (mod) 10469 has been embodied in production; and ATR 72-101, ATR 72-102, ATR 72-201, ATR 72-202, ATR 72-211, ATR 72-212 and ATR 72-212A aeroplanes, all MSN, except those on which ATR mod 10453 has been embodied in production.		
主旨摘要 Subject	Flight Controls - Rudder Travel Limitation Unit Logic - Modification		
民航局 CAA <input type="radio"/> 本國產品 Native product <input type="radio"/> 其他個案 Other	設計國民航主管機構 Original Authority <input type="radio"/> FAA <input type="radio"/> Germany LBA <input checked="" type="radio"/> EASA <input type="radio"/> CAA-NL <input type="radio"/> Brazil <input type="radio"/> UK CAA <input type="radio"/> Transport Canada Civil Aviation <input type="radio"/> Japan CAB <input type="radio"/> DGAC <input type="radio"/> CAA of Israel <input type="radio"/> Other_____		
	設計國AD編號 Original AD number	2025-0004	
	1. 直接採用原AD之內容? (Is the original AD directly adopted?) <input checked="" type="radio"/> 是(Yes) <input type="radio"/> 否(No)_ a. 生效日期另訂為(Re-specify the effective date as): b. 執行時限另訂為(Re-specify the compliance time or period as): 2. 使用人是否需要將AD執行結果向民航局提出報告? (Do users need to report the status of compliance to the CAA?) <input type="radio"/> 需要(Yes) <input checked="" type="radio"/> 不需要(No)		
備註 Note	ATA 27. Ref. Publications: ATR SB ATR42-27-0115 revision 01 dated 29 November 2024. and ATR SB ATR42-27-0118 revision 01 dated 29 November 2024. and ATR SB ATR72-27-1075 revision 01 dated 29 November 2024. and ATR SB ATR72-27-1077 revision 02 dated 29 November 2024.		

註： 1. AD內容後附。

2. 航空器產品使用人得向民航局提出豁免、替代符合方法、執行時限之展延之申請。

3. 如有任何問題，請聯絡交通部民用航空局初始適航科。Tel：(02)2349-6330 / 6332, Fax：(02)2545-8464, adcaa@mail.caa.gov.tw

Note： 1. The AD text is enclosed.

2. Exemption, an alternative method of compliance or adjustment of the compliance time may be proposed to the CAA for approval.

3. For further information, please contact Civil Aviation Administration on Tel：(02)2349-6330 / 6332, Fax：(02)2545-8464, adcaa@mail.caa.gov.tw



Airworthiness Directive

AD No.: 2025-0004

Issued: 07 January 2025

Note: This Airworthiness Directive (AD) is issued by EASA, acting in accordance with Regulation (EU) 2018/1139 on behalf of the European Union, its Member States and of the European third countries that participate in the activities of EASA under Article 129 of that Regulation.

This AD is issued in accordance with Regulation (EU) 748/2012, Part 21.A.3B. In accordance with Regulation (EU) 1321/2014 Annex I Part M.A.301, or Annex Vb Part ML.A.301, as applicable, the continuing airworthiness of an aircraft shall be ensured by accomplishing any applicable ADs. Consequently, no person may operate an aircraft to which an AD applies, except in accordance with the requirements of that AD, unless otherwise specified by the Agency [Regulation (EU) 1321/2014 Annex I Part M.A.303, or Annex Vb Part ML.A.303, as applicable] or agreed with the Authority of the State of Registry [Regulation (EU) 2018/1139, Article 71 exemption].

Design Approval Holder's Name:

ATR-GIE AVIONS DE TRANSPORT REGIONAL

Type/Model designation(s):

ATR 42 and ATR 72 aeroplanes

Effective Date: 21 January 2025

TCDS Number(s): EASA.A.084

Foreign AD: Not applicable

Supersedure: None

ATA 27 – Flight Controls – Rudder Travel Limitation Unit Logic – Modification

Manufacturer(s):

ATR-GIE Avions de Transport Régional, formerly EADS ATR - Alenia, Aerospatiale Matra ATR - ALENIA, Aerospatiale - Alenia, Aerospatiale - Aeritalia

Applicability:

ATR 42-400 and ATR 42-500 aeroplanes, all manufacturer serial numbers (MSN), except those on which ATR modification (mod) 10469 has been embodied in production; and
ATR 72-101, ATR 72-102, ATR 72-201, ATR 72-202, ATR 72-211, ATR 72-212 and ATR 72-212A aeroplanes, all MSN, except those on which ATR mod 10453 has been embodied in production.

Definitions:

For the purpose of this AD, the following definitions apply:

The applicable SB1: ATR Service Bulletin (SB) ATR42-27-0115 revision 01 or ATR72-27-1075 revision 01, as applicable.

The applicable SB2: ATR SB ATR42-27-0118 revision 01 or ATR72-27-1077 revision 02, as applicable.



Groups: Group 1 aeroplanes are:

ATR 42-400 and 42-500 all MSN which do not have ATR mod 05948 embodied.

ATR 72-101, ATR 72-102, ATR 72-201, ATR 72-202, ATR 72-211, ATR 72-212 and ATR 72-212A, all MSN which do not have ATR mod 05948 embodied.

Group 2 aeroplanes are:

ATR 42-400 and 42-500 all MSN which have ATR mod 05948 embodied.

ATR 72-101, ATR 72-102, ATR 72-201, ATR 72-202, ATR 72-211, ATR 72-212 and ATR 72-212A, all MSN which have ATR mod 05948 embodied.

Reason:

During an ATR internal review of the Travel Limitation Unit (TLU) new avionics suite design, it has been identified that an erroneous behaviour of Core Processing Module (CPM) 2, hosting the Data Concentration Application (DCA) 2, might affect the TLU command, monitoring and indication. Further investigation results indicated that an erroneous monitoring of the TLU could occur when the aeroplane is flying above 185 kt (ATR72) or 195 kt (ATR42), due to the fact that the logic takes into account either ADC1 or ADC2 input.

This condition, if not corrected, could result in the rudder deflection not being limited at high aeroplane speed which, if combined with a large rudder pedal input, could lead to the loss of control of the aeroplane.

To address this potential unsafe condition, ATR developed mod 10469 (for ATR42 aeroplanes) and mod 10453 (for ATR72 aeroplanes), and issued the applicable SB1 and SB2 to provide modification instructions for in-service aeroplanes.

During the PAD consultation period it has been identified that, after changes made in accordance with the instructions of the SB ATR72-27-1075 at original issue, SB ATR72-27-1077 at original issue or at revision 01, SB ATR42-27-0115 at original issue, or SB ATR42-27-0118 at original issue, as applicable, the Maintenance Procedure (MP) task ATR-A-27-23-XX-00ZZZ-340Z-A cannot be successfully accomplished. Consequently, ATR revised the above-mentioned SBs and revised the MP ATR-A-27-23-XX-00ZZZ-340Z-A.

For the reasons described above, this AD requires installation of one or two relay(s) and associated wiring to improve the TLU command and monitoring logics, considering both ADC1 and ADC2 inputs and requires additional work for certain aeroplanes.

Required Action(s) and Compliance Time(s):

Required as indicated by this AD, unless the action(s) required by this AD have been already accomplished:

Modification:

- (1) Within the compliance time as defined in Table 1 of this AD, as applicable, modify the aeroplane in accordance with the instructions of section 3.C of the applicable SB.



Table 1 – Modification Compliance Time

Group	Compliance Time (after the effective date of this AD)	Applicable SB
1	Within 24 months	Applicable SB1
2	Within 12 months	Applicable SB2

Tests:

- (2) Before next flight after the modification of an aeroplane as required by paragraph (1) of this AD, accomplish the tests in accordance with the instructions of section 3.D of the applicable SB.

Corrective Action(s):

- (3) If, during the accomplishment of the tests, as required by paragraph (2) of this AD, any discrepancy is detected, before next flight contact ATR for applicable instructions and accomplish those instructions accordingly.

Note 1: Following modification of an aeroplane as required by paragraph (1) of this AD, that aeroplane is considered to be in post-mod 10469 (ATR42) or post-mod 10453 (ATR72) configuration. ATR issued Time Limits Document (TLD) temporary revisions introducing applicable maintenance requirements, which are referenced in EASA AD 2025-0005 and EASA AD 2025-0006.

Additional Work:

- (4) For Group 1 and 2 aeroplanes that, before the effective date of this AD, have been modified in accordance with the instructions of ATR SB ATR72-27-1075 at original issue, SB ATR72-27-1077 at original issue or at revision 01, or SB ATR42-27-0115 at original issue, or SB ATR42-27-0118 at original issue: Within 750 flight hours or 6 months, whichever occurs first after the effective date of this AD, accomplish the additional work as identified in, and in accordance with the instructions of, the SB 1 or the SB 2 as applicable.

Ref. Publications:

ATR SB ATR42-27-0115 revision 01 dated 29 November 2024.

ATR SB ATR42-27-0118 revision 01 dated 29 November 2024.

ATR SB ATR72-27-1075 revision 01 dated 29 November 2024.

ATR SB ATR72-27-1077 revision 02 dated 29 November 2024.

The use of later approved revisions of the above-mentioned documents is acceptable for compliance with the requirements of this AD.

Remarks:

1. If requested and appropriately substantiated, EASA can approve Alternative Methods of Compliance for this AD.



2. This AD was posted on 25 October 2024 as PAD 24-129 for consultation until 22 November 2024 and republished on 10 December 2024 as PAD 24-129R1 for additional consultation until 03 January 2025. The Comment Response Documents can be found in the [EASA Safety Publications Tool](#), in the compressed ('zipped') file, attached to the record for this AD.
3. Enquiries regarding this AD should be referred to the EASA Safety Information Section, Certification Directorate. E-mail: ADs@easa.europa.eu.
4. Information about any failures, malfunctions, defects or other occurrences, which may be similar to the unsafe condition addressed by this AD, and which may occur, or have occurred on a product, part or appliance not affected by this AD, can be reported to the [EU aviation safety reporting system](#). This may include reporting on the same or similar components, other than those covered by the design to which this AD applies, if the same unsafe condition can exist or may develop on an aircraft with those components installed. Such components may be installed under an FAA Parts Manufacturer Approval (PMA), Supplemental Type Certificate (STC) or other modification.
5. For any question concerning the technical content of the requirements in this AD, please contact: ATR - GIE Avions de Transport Régional, Continued Airworthiness Service, Telephone: +33 (0)5 62 21 62 21, Fax: +33 (0) 5 62 21 67 18; E-mail: continued.airworthiness@atr-aircraft.com.

