



# 適航指令發布單

## Airworthiness Directive Issuance Form

民航局AD編號 AD number	CAA-2024-10-012	發布日期 Date issued	2024/10/25
適用之航空產品 Applied to (models, serial numbers or part numbers, as applicable)	Airbus A319-111, A319-112, A319-113, A319-114, A319-115, A319-131, A319-132, A319-133, A320-211, A320-212, A320-214, A320-215, A320-216, A320-231, A320-232, A320-233, A320-251N, A320-252N, A320-253N, A320-271N, A320-272N, A320-273N, A321-211, A321-212, A321-213, A321-231, A321-232, A321-251N, A321-251NX, A321-252N, A321-252NX, A321-253N, A321-253NX, A321-253NY, A321-271N, A321-271NX, A321-272N and A321-272NX aeroplanes; and Airbus A330-201, A330-202, A330-203, A330-223, A330-243, A330-301, A330-302, A330-303, A330-321, A330-322, A330-323, A330-341, A330-342, A330-343, A330-841 and A330-941 aeroplanes; and Airbus A350-941 and A350-1041 aeroplanes; and Airbus A380-841, A380-842 and A380-861 aeroplanes; all manufacturer serial numbers.		
主旨摘要 Subject	Cabin Systems - Antenna Adapter Plate - Inspection		
民航局 CAA  <input type="radio"/> 本國產品 Native product  <input type="radio"/> 其他個案 Other	設計國民航主管機構 Original Authority  <input type="radio"/> FAA <input type="radio"/> Germany LBA <input checked="" type="radio"/> EASA <input type="radio"/> CAA-NL <input type="radio"/> Brazil <input type="radio"/> UK CAA <input type="radio"/> Transport Canada Civil Aviation <input type="radio"/> Japan CAB <input type="radio"/> DGAC <input type="radio"/> CAA of Israel <input type="radio"/> Other_____		
	設計國AD編號 Original AD number	2024-0199	
	1. 直接採用原AD之內容? (Is the original AD directly adopted?) <input checked="" type="radio"/> 是(Yes) <input type="radio"/> 否(No)_ a. 生效日期另訂為(Re-specify the effective date as): b. 執行時限另訂為(Re-specify the compliance time or period as): 2. 使用人是否需要將AD執行結果向民航局提出報告? (Do users need to report the status of compliance to the CAA?) <input type="radio"/> 需要(Yes) <input checked="" type="radio"/> 不需要(No)		

備註  
Note

This AD supersedes EASA AD 2024-0106(CAA-2024-06-001)  
dated 31 May 2024.

- 註：
1. AD內容後附。
  2. 航空器產品使用人得向民航局提出豁免、替代符合方法、執行時限之展延之申請。
  3. 如有任何問題，請聯絡交通部民用航空局初始適航科。Tel：(02)2349-6330 / 6332, Fax：(02)2545-8464, [adcaa@mail.caa.gov.tw](mailto:adcaa@mail.caa.gov.tw)

- Note：
1. The AD text is enclosed.
  2. Exemption, an alternative method of compliance or adjustment of the compliance time may be proposed to the CAA for approval.
  3. For further information, please contact Civil Aviation Administration on Tel：(02)2349-6330 / 6332, Fax：(02)2545-8464, [adcaa@mail.caa.gov.tw](mailto:adcaa@mail.caa.gov.tw)



## Airworthiness Directive

**AD No.:** 2024-0199

**Issued:** 18 October 2024

Note: This Airworthiness Directive (AD) is issued by EASA, acting in accordance with Regulation (EU) 2018/1139 on behalf of the European Union, its Member States and of the European third countries that participate in the activities of EASA under Article 129 of that Regulation.

This AD is issued in accordance with Regulation (EU) 748/2012, Part 21.A.3B. In accordance with Regulation (EU) 1321/2014 Annex I Part M.A.301, or Annex Vb Part ML.A.301, as applicable, the continuing airworthiness of an aircraft shall be ensured by accomplishing any applicable ADs. Consequently, no person may operate an aircraft to which an AD applies, except in accordance with the requirements of that AD, unless otherwise specified by the Agency [Regulation (EU) 1321/2014 Annex I Part M.A.303, or Annex Vb Part ML.A.303, as applicable] or agreed with the Authority of the State of Registry [Regulation (EU) 2018/1139, Article 71 exemption].

### Design Approval Holder's Name:

AIRBUS S.A.S.

### Type/Model designation(s):

A319, A320, A321, A330, A350 and A380 aeroplanes

**Effective Date:** 01 November 2024

**TCDS Number(s):** EASA.A.064

**Foreign AD:** Not applicable

**Supersedure:** This AD supersedes EASA AD 2024-0106 dated 31 May 2024.

## ATA 44 – Cabin Systems – Antenna Adapter Plate – Inspection

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### Manufacturer(s):

Airbus, formerly Airbus Industrie

### Applicability:

Airbus A319-111, A319-112, A319-113, A319-114, A319-115, A319-131, A319-132, A319-133, A320-211, A320-212, A320-214, A320-215, A320-216, A320-231, A320-232, A320-233, A320-251N, A320-252N, A320-253N, A320-271N, A320-272N, A320-273N, A321-211, A321-212, A321-213, A321-231, A321-232, A321-251N, A321-251NX, A321-252N, A321-252NX, A321-253N, A321-253NX, A321-253NY, A321-271N, A321-271NX, A321-272N and A321-272NX aeroplanes; and

Airbus A330-201, A330-202, A330-203, A330-223, A330-243, A330-301, A330-302, A330-303, A330-321, A330-322, A330-323, A330-341, A330-342, A330-343, A330-841 and A330-941 aeroplanes; and

Airbus A350-941 and A350-1041 aeroplanes; and

Airbus A380-841, A380-842 and A380-861 aeroplanes;

all manufacturer serial numbers.



**Definitions:**

For the purpose of this AD, the following definitions apply:

**Affected parts:** Broadband antenna adapter plate, skirt, vents and attachment fittings installed, or eligible for installation on an aeroplane, by embodiment of Airbus production modification (MOD) as specified in Appendix 1 of this AD, or in accordance with the instructions of Airbus Service Bulletin (SB) listed in Appendix 1 of this AD.

**Aeroplane date of manufacture:** The date of transfer of title (ownership) of the aeroplane upon delivery by Airbus to the first operator, which is referenced in Airbus documentation.

**Panasonic SB:** Panasonic Avionics Corporation (Panasonic) SB RDAA903008-01-44-00, SB RDAA903141-01-44-00, SB RDAA903194-01-44-00, SB RDAA903366-01-44-00, SB RDAA903465-01-44-00 or SB RDAA903528-01-44-00, as applicable.

**The applicable SB:** Airbus SB A320-44-1102, SB A320-44-1103, SB A330-44-3054, SB A350-44-P090 or SB A380-44-8145, as applicable to aeroplane model, and referencing the applicable Panasonic SB.

**Groups:** Group 1 aeroplanes are those that have an affected part installed.

Group 2 aeroplanes are those that do not have an affected part installed.

An A321-253NY aeroplane is a Group 2 aeroplane, provided that no affected part has been installed on that aeroplane since aeroplane date of manufacture.

**Reason:**

Corrosion and cracks on the affected parts have been reported after an inspection. Further investigation determined that the affected parts are made of material susceptible to corrosion cracking, and that the recommended maintenance programs do not ensure timely detection of cracks or damages in this area.

This condition, if not detected and corrected, could lead to in-flight detachment of radome, antenna and/or affected parts, which could impact the tail section of the aeroplane, possibly resulting in damage to, and reduced control of, the aeroplane.

To address this potential unsafe condition, Airbus issued the applicable SB, which refer to SB issued by Panasonic (current manufacturer of the affected part), to provide inspection instructions for the affected parts. Consequently, EASA issued AD 2024-0106 to require repetitive general visual inspections (GVI) of the affected parts, and, depending on findings, accomplishment of applicable corrective action(s).

Since that AD was issued, a new aeroplane model (A321-253NY) has been certified, on which affected parts are eligible for installation. On the issue date of this AD, no A321-253NY aeroplanes have been delivered yet to operators. Furthermore, new Revisions of the Panasonic SB were issued and certain previously quoted revisions have been removed from the Ref. Publications section, since they were never dispatched to operators.



For the reason described above, this AD supersedes EASA AD 2024-0106, extends the Applicability to the A321-253NY aeroplanes and updates the list of the Panasonic SB in the Ref. Publication section.

For aeroplanes previously affected by EASA AD 2024-0106, this AD retains the requirements of that AD, with no additional actions.

This AD is still considered to be an interim action and further AD action may follow.

#### **Required Action(s) and Compliance Time(s):**

Required as indicated by this AD, unless the action(s) required by this AD have been already accomplished:

#### **Inspections:**

- (1) For Group 1 aeroplanes: Within the compliance time as specified in Appendix A of the applicable SB, and, thereafter, at intervals not exceeding 36 months, accomplish a GVI of the affected parts in accordance with the instructions of the Panasonic SB, as referenced in the applicable SB.

Where the applicable SB refers to “years since date of Antenna system embodiment”, the years accumulated on 14 June 2024 [the effective date of EASA AD 2024-0106] since the date of first installation of that affected part on that aeroplane must be used instead; if the affected part was installed during aircraft production or, if the date of first installation is unknown, the aeroplane date of manufacture must be used instead. Where the applicable SB refers to the “SB publication date”, 14 June 2024 [the effective date of EASA AD 2024-0106] must be used instead.

#### **Corrective Action(s):**

- (2) If, during any GVI as required by paragraph (1) of this AD, any crack and/or corrosion are detected on an affected part, before next flight, accomplish the applicable corrective action(s) in accordance with the instructions of the applicable SB, or contact Airbus for approved repair instructions and accomplish those instructions accordingly, or replace that part with a serviceable part (see paragraph (5) of this AD).

#### **Terminating Action:**

- (3) None.

#### **Reporting:**

- (4) Within 30 days after each GVI as required by paragraph (1) of this AD, report the inspection results (including no findings for the first inspection only) to Airbus. Using the inspection report attached to the applicable SB is an acceptable method to comply with this requirement.

#### **Part(s) Installation:**

- (5) For Group 1 and Group 2 aeroplanes: From 14 June 2024 [the effective date of EASA AD 2024-0106], it is allowed to install an affected part on an aeroplane, provided it is new or that it has accumulated less than 36 months since previous GVI, or that, before next flight after installation, it is inspected and, depending on findings, corrective action are accomplished as



required by paragraphs (1) and (2) of this AD (see Note 1 of this AD).

After installation of affected part on a Group 2 aeroplane, the aeroplane is effectively considered a Group 1 and must be inspected as required by this AD.

Note 1: Removal of an affected part and reinstallation of an affected part on the same aeroplane, accomplished during a single maintenance visit, is not considered as installation as specified in paragraph (5) of this AD.

**Credit:**

- (6) Inspection(s) and corrective action(s), accomplished on an aeroplane, before 14 June 2024 [the effective date of EASA AD 2024-0106], in accordance with the instructions of Airbus SB A380-44-8145 original issue are acceptable to comply with the initial requirements of paragraph (1) or (2) of this AD, as applicable, for that aeroplane.

**Alternative Method of Compliance:**

- (7) Inspection(s) accomplished on an aeroplane, in accordance with the instructions of the following Maintenance Procedures are acceptable to comply with the requirements of paragraph (1) of this AD, for that aeroplane:

A320: TASK 44-33-21-400-801-A - Installation of the Adapter Plate Kit.

A330: TASK 44-33-21-400-801-A - Installation of the Adapter Plate-Kit.

A350: TASK A350-A-44-33-11-A7001-720A-A - Installation of the Broadband Antenna Adapter-Plate.

A380: TASK 44-33-25-400-801-A - Installation of the Adapter Plate Kit.

**Ref. Publications:**

Airbus SB A320-44-1102 original issue dated 22 December 2023.

Airbus SB A320-44-1103 original issue dated 22 December 2023.

Airbus SB A330-44-3054 original issue dated 22 December 2023, or Revision 01 dated 11 July 2024.

Airbus SB A350-44-P090 original issue dated 22 December 2023.

Airbus SB A380-44-8145 revision 1 dated 22 April 2024.

Panasonic SB RDAA903008-01-44-00 Revision 1 dated 14 December 2023 and Revision 2 dated 23 April 2024.

Panasonic SB RDAA903141-01-44-00 Revision 1 dated 18 December 2023 and Revision 02 dated 23 April 2024.



Panasonic SB RDAA903194-01-44-00 Revision 1 dated 14 December 2023 and Revision 02 dated 03 May 2024.

Panasonic SB RDAA903366-01-44-00 Revision 2 dated 21 December 2023 and Revision 3 dated 03 May 2024.

Panasonic SB RDAA903465-01-44-00 Revision 1 dated 19 December 2023 and Revision 2 dated 23 April 2024.

Panasonic SB RDAA903528-01-44-00 Revision 1 dated 19 December 2023 and Revision 02 dated 22 April 2024.

The use of later approved revisions of the above-mentioned documents is acceptable for compliance with the requirements of this AD.

#### Remarks:

1. If requested and appropriately substantiated, EASA can approve Alternative Methods of Compliance for this AD.
2. Based on the required actions and the compliance time, EASA have decided to issue a Final AD with Request for Comments, postponing the public consultation process until after publication. All interested persons may send their comments, referencing the AD Number, to the E-mail address specified in below Remark 3, prior to 15 November 2024. Only if any comment is received during the consultation period, a Comment Response Document will be published in the [EASA Safety Publications Tool](#), in a compressed ('zipped') file, attached to the record for this AD.
3. Enquiries regarding this AD should be referred to the EASA Safety Information Section, Certification Directorate. E-mail: [ADs@easa.europa.eu](mailto:ADs@easa.europa.eu).
4. Information about any failures, malfunctions, defects or other occurrences, which may be similar to the unsafe condition addressed by this AD, and which may occur, or have occurred on a product, part or appliance not affected by this AD, can be reported to the [EU aviation safety reporting system](#). This may include reporting on the same or similar components, other than those covered by the design to which this AD applies, if the same unsafe condition can exist or may develop on an aircraft with those components installed. Such components may be installed under an FAA Parts Manufacturer Approval (PMA), Supplemental Type Certificate (STC) or other modification.
5. For any question concerning the technical content of the requirements in this AD, please contact: AIRBUS, as specified below:

For A320 family program: Airworthiness Office,  
E-mail: [account.airworth-eas@airbus.com](mailto:account.airworth-eas@airbus.com).

For A330 family program: Airworthiness Office,  
E-mail: [airworthiness.A330-A340@airbus.com](mailto:airworthiness.A330-A340@airbus.com).



For A350 family program: Airworthiness Office,  
E-mail: [continued-airworthiness.a350@airbus.com](mailto:continued-airworthiness.a350@airbus.com).

For A380 family program: Airworthiness Office,  
E-mail: [account.airworth-A380@airbus.com](mailto:account.airworth-A380@airbus.com).



**Appendix 1 – List of MOD/SB Installing Affected Parts**

<b>Production MOD</b>	<b>Program Applicability</b>	<b>Retrofit SB</b>	<b>KIT Part Number</b>	<b>Designation</b>
100549 and any of 113406 or 115826 or 108765	A350	-	RD-AA903366-01 RDAA903366-01*	TOP KIT - Antenna System, PAC GCS, A350 Single Panel antenna (SPA)
100549 and none of 113406 or 115826 or 108765	A350	-	RD-AA903008-01 RDAA903008-01*	TOP KIT - Antenna System, PAC GCS, A350 Dual Panel Antenna (DPA)
73057	A380	A380-44-8112 or A380-44-8113	RD-AA903141-01 RDAA903141-01*	Adapter Plate Installation Kit DPA (A380)
203251 or 204608	A330	A330-44-3025	RD-AA903194-01 RDAA903194-01*	Adapter Plate Installation Kit DPA (A330)
207191	A330	-	RD-AA903528-01 RDAA903528-01*	Adapter Plate Installation Kit SPA (A330)
158784 or 160690	A320	-	RD-AA903465-01 RDAA903465-01*	Adapter Plate Installation Kit SPA (A320 family)

\* PN as mentioned in Reference Publication Airbus' SBs

