



# 適航指令發布單

## Airworthiness Directive Issuance Form

民航局AD編號 AD number	CAA-2024-10-006修訂	發布日期 Date issued	2024/10/17
適用之航空產品 Applied to (models, serial numbers or part numbers, as applicable)	Airbus A350-941 and A350-1041 aeroplanes; and Airbus A380-841 and A380-842 aeroplanes; all manufacturer serial numbers.		
主旨摘要 Subject	Aircraft Flight Manual - Normal Procedures - Amendment		
民航局 CAA  <input type="radio"/> 本國產品 Native product  <input type="radio"/> 其他個案 Other	設計國民航主管機構 Original Authority  <input type="radio"/> FAA <input type="radio"/> Germany LBA <input checked="" type="radio"/> EASA <input type="radio"/> CAA-NL <input type="radio"/> Brazil <input type="radio"/> UK CAA <input type="radio"/> Transport Canada Civil Aviation <input type="radio"/> Japan CAB <input type="radio"/> DGAC <input type="radio"/> CAA of Israel <input type="radio"/> Other_____		
	設計國AD編號 Original AD number	2024-0190Correction	
	1. 直接採用原AD之內容? (Is the original AD directly adopted?) <input checked="" type="radio"/> 是(Yes) <input type="radio"/> 否(No)_ a. 生效日期另訂為(Re-specify the effective date as): b. 執行時限另訂為(Re-specify the compliance time or period as): 2. 使用人是否需要將AD執行結果向民航局提出報告? (Do users need to report the status of compliance to the CAA?) <input type="radio"/> 需要(Yes) <input checked="" type="radio"/> 不需要(No)		
備註 Note	This AD is republished to correct the revision of FAA AD 2024-0190(CAA-2024-10-006)。		

註： 1. AD內容後附。  
2. 航空器產品使用人得向民航局提出豁免、替代符合方法、執行時限之展延之申請。  
3. 如有任何問題，請聯絡交通部民用航空局初始適航科。Tel：(02)2349-6330 / 6332, Fax：(02)2545-8464, [adcaa@mail.caa.gov.tw](mailto:adcaa@mail.caa.gov.tw)

Note： 1. The AD text is enclosed.  
2. Exemption, an alternative method of compliance or adjustment of the compliance time may be proposed to the CAA for approval.  
3. For further information, please contact Civil Aviation Administration on Tel：(02)2349-6330 / 6332, Fax：(02)2545-8464, [adcaa@mail.caa.gov.tw](mailto:adcaa@mail.caa.gov.tw)



## Airworthiness Directive

**AD No.:** 2024-0190

**Issued:** 10 October 2024

**[Correction:** 11 October 2024]

Note: This Airworthiness Directive (AD) is issued by EASA, acting in accordance with Regulation (EU) 2018/1139 on behalf of the European Union, its Member States and of the European third countries that participate in the activities of EASA under Article 129 of that Regulation.

This AD is issued in accordance with Regulation (EU) 748/2012, Part 21.A.3B. In accordance with Regulation (EU) 1321/2014 Annex I Part M.A.301, or Annex Vb Part ML.A.301, as applicable, the continuing airworthiness of an aircraft shall be ensured by accomplishing any applicable ADs. Consequently, no person may operate an aircraft to which an AD applies, except in accordance with the requirements of that AD, unless otherwise specified by the Agency [Regulation (EU) 1321/2014 Annex I Part M.A.303, or Annex Vb Part ML.A.303, as applicable] or agreed with the Authority of the State of Registry [Regulation (EU) 2018/1139, Article 71 exemption].

**Design Approval Holder's Name:**

AIRBUS S.A.S.

**Type/Model designation(s):**

A350 and A380 aeroplanes

**Effective Date:** 24 October 2024

**TCDS Number(s):** EASA.A.151 and EASA.A.110

**Foreign AD:** Not applicable

**Supersedure:** None

### ATA – Aircraft Flight Manual – Normal Procedures – Amendment

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**Manufacturer(s):**

Airbus

**Applicability:**

Airbus A350-941 and A350-1041 aeroplanes; and  
Airbus A380-841 and A380-842 aeroplanes;  
all manufacturer serial numbers.

**Definitions:**

For the purpose of this AD, the following definitions apply:

**The AFM DU revision:**

Airbus A350 Airplane Flight Manual (AFM) Documentary Unit (DU):

- 00017714.0001001 dated 19 January 2024; or
- 00017714.0002001 dated 19 January 2024; or
- 00017714.0003001 dated 06 February 2024; or

Airbus A380 AFM DU:

- 00015818.0001001 dated 11 July 2024,

as applicable to aeroplane model and depending on aeroplane configuration.



**Reason:**

It has been determined that, for A350 and A380 aeroplanes equipped with Rolls-Royce engines, the applicable AFM was providing an incorrect value for maximum cumulative taxi time in freezing fog conditions.

This condition, if not corrected, could lead to multiple engine surge in a critical flight phase, possibly resulting in loss of control of the aeroplane.

To address this potential unsafe condition, Airbus published the AFM DU revision, as defined in this AD, correcting the maximum cumulative taxi time in freezing fog conditions.

For the reasons described above, this AD requires amendment of the applicable AFM DU.

This AD is republished to correct the revision date of the A350 AFM in paragraph (2).

**Required Action(s) and Compliance Time(s):**

Required as indicated by this AD, unless the action(s) required by this AD have been already accomplished:

**AFM Amendment:**

- (1) Within 1 month after the effective date of this AD, implement the AFM DU revision, as defined in this AD, inform all flight crews, and thereafter, operate the aeroplane accordingly.
- (2) Amending the AFM of an aeroplane by incorporating Airbus A350 AFM revision dated 06 February 2024 or Airbus A380 AFM revision dated 11 July 2024, as applicable, or later revision, is an acceptable method to comply with the requirements of paragraph (1) of this AD for that aeroplane.

**Ref. Publications:**

Airbus A350 AFM DU 00017714.0001001 dated 19 January 2024.

Airbus A350 AFM DU 00017714.0002001 dated 19 January 2024.

Airbus A350 AFM DU 00017714.0003001 dated 06 February 2024.

Airbus A380 AFM DU 00015818.0001001 dated 11 July 2024.

The use of later approved revisions of the above-mentioned documents is acceptable for compliance with the requirements of this AD.

**Remarks:**

1. If requested and appropriately substantiated, EASA can approve Alternative Methods of Compliance for this AD.
2. Based on the required actions and the compliance time, EASA have decided to issue a Final AD with Request for Comments, postponing the public consultation process until after publication.



All interested persons may send their comments, referencing the AD Number, to the E-mail address specified in below Remark 3, prior to 07 November 2024. Only if any comment is received during the consultation period, a Comment Response Document will be published in the [EASA Safety Publications Tool](#), in a compressed ('zipped') file, attached to the record for this AD.

3. Enquiries regarding this AD should be referred to the EASA Safety Information Section, Certification Directorate. E-mail: [ADs@easa.europa.eu](mailto:ADs@easa.europa.eu).
4. Information about any failures, malfunctions, defects or other occurrences, which may be similar to the unsafe condition addressed by this AD, and which may occur, or have occurred on a product, part or appliance not affected by this AD, can be reported to the [EU aviation safety reporting system](#). This may include reporting on the same or similar components, other than those covered by the design to which this AD applies, if the same unsafe condition can exist or may develop on an aircraft with those components installed. Such components may be installed under an FAA Parts Manufacturer Approval (PMA), Supplemental Type Certificate (STC) or other modification.
5. For any question concerning the technical content of the requirements in this AD, please contact: AIRBUS, as specified below.

For A350 family program: Airworthiness Office,  
e-mail: [continued-airworthiness.a350@airbus.com](mailto:continued-airworthiness.a350@airbus.com)

For A380 family program: Airworthiness Office,  
e-mail: [account.airworth-A380@airbus.com](mailto:account.airworth-A380@airbus.com)

