



# 適航指令發布單

## Airworthiness Directive Issuance Form

民航局AD編號 AD number	CAA-2024-02-011	發布日期 Date issued	2024/02/19
適用之航空產品 Applied to (models, serial numbers or part numbers, as applicable)	A321-251NX, A321-252NX, A321-253NX, A321-271NX and A321-272NX aeroplanes, all manufacturer serial numbers.		
主旨摘要 Subject	Equipment/Furnishings - Escape Slide and Offwing Slide Inflation Reservoir Venting Holes - Inspection / Replacement		
民航局 CAA	設計國民航主管機構 Original Authority		
<input type="radio"/> 本國產品 Native product	<input type="radio"/> FAA	<input type="radio"/> Germany LBA	
<input type="radio"/> 其他個案 Other	<input checked="" type="radio"/> EASA	<input type="radio"/> CAA-NL	
	<input type="radio"/> Brazil	<input type="radio"/> UK CAA	
	<input type="radio"/> Transport Canada Civil Aviation	<input type="radio"/> Japan CAB	
	<input type="radio"/> DGAC	<input type="radio"/> CAA of Israel	
		<input type="radio"/> Other_____	
	設計國AD編號 Original AD number	2024-0048	
	1. 直接採用原AD之內容? (Is the original AD directly adopted?) <input checked="" type="radio"/> 是(Yes) <input type="radio"/> 否(No)_ a. 生效日期另訂為(Re-specify the effective date as) : b. 執行時限另訂為(Re-specify the compliance time or period as) :		
	2. 使用人是否需要將AD執行結果向民航局提出報告? (Do users need to report the status of compliance to the CAA?) <input type="radio"/> 需要(Yes) <input checked="" type="radio"/> 不需要(No)		
備註 Note	ATA 25. Ref. Publications: Airbus AOT 25N027-23 original issue dated 16 January 2024.		

註： 1. AD內容後附。

2. 航空器產品使用人得向民航局提出豁免、替代符合方法、執行時限之展延之申請。

3. 如有任何問題，請聯絡交通部民用航空局初始適航科。Tel：(02)2349-6330 / 6332, Fax：(02)2545-8464, [adcaa@mail.caa.gov.tw](mailto:adcaa@mail.caa.gov.tw)

Note： 1. The AD text is enclosed.

2. Exemption, an alternative method of compliance or adjustment of the compliance time may be proposed to the CAA for approval.

3. For further information, please contact Civil Aeronautics Administration on Tel：(02)2349-6330 / 6332, Fax：(02)2545-8464, [adcaa@mail.caa.gov.tw](mailto:adcaa@mail.caa.gov.tw)



## Airworthiness Directive

**AD No.:** 2024-0048

**Issued:** 16 February 2024

Note: This Airworthiness Directive (AD) is issued by EASA, acting in accordance with Regulation (EU) 2018/1139 on behalf of the European Union, its Member States and of the European third countries that participate in the activities of EASA under Article 129 of that Regulation.

This AD is issued in accordance with Regulation (EU) 748/2012, Part 21.A.3B. In accordance with Regulation (EU) 1321/2014 Annex I Part M.A.301, or Annex Vb Part ML.A.301, as applicable, the continuing airworthiness of an aircraft shall be ensured by accomplishing any applicable ADs. Consequently, no person may operate an aircraft to which an AD applies, except in accordance with the requirements of that AD, unless otherwise specified by the Agency [Regulation (EU) 1321/2014 Annex I Part M.A.303, or Annex Vb Part ML.A.303, as applicable] or agreed with the Authority of the State of Registry [Regulation (EU) 2018/1139, Article 71 exemption].

### Design Approval Holder's Name:

AIRBUS S.A.S.

### Type/Model designation(s):

A321 aeroplanes

**Effective Date:** 01 March 2024

**TCDS Number(s):** EASA.A.064

**Foreign AD:** Not applicable

**Supersedure:** None

## ATA 25 – Equipment/Furnishings – Escape Slide and Offwing Slide Inflation Reservoir Venting Holes – Inspection / Replacement

### Manufacturer(s):

Airbus, formerly Airbus Industries

### Applicability:

A321-251NX, A321-252NX, A321-253NX, A321-271NX and A321-272NX aeroplanes, all manufacturer serial numbers.

### Definitions:

For the purpose of this AD, the following definitions apply:

**The AOT:** Airbus Alert Operators Transmission (AOT) A25N027-23.

**The VSB:** Safran Service Bulletin (SB) 005-25-45 or SB 005-25-46, as applicable.

**Affected part:** Any orifice fitting having Part Number (P/N) M3SP-303-004-E installed on a reservoir having a P/N as defined in Annex 1 of this AD and a serial number as identified in the VSB, except those that have been modified in accordance with the instructions of the AOT or the VSB.



**Serviceable part:** Any orifice fitting, eligible for installation in accordance with instructions approved by Airbus, that is not an affected part.

**Groups:** Group 1 aeroplanes are those which have an affected part installed. Group 2 aeroplanes are those which are not Group 1 aeroplanes.

**Reason:**

Following a quality review during manufacturing, a quality escape was identified on A321NX Door 3 and Offwing inflation reservoirs' venting holes, where a torque strip indicator (material "Dykem") has been applied on the orifice fitting.

This condition, in combination with a slide reservoir pressure loss, if not detected and corrected, could lead to deployment in flight of a non-inflated slide, possibly resulting in damage to, and reduced control of, the aeroplane.

To address this potential unsafe condition, Airbus issued the AOT, which refers to the VSB, to provide instructions for inspection and corrective action(s) of the affected parts.

For the reasons described above, this AD requires a one-time general visual inspection (GVI) of the affected parts, and, in case of findings, replacement of the orifice fitting.

**Required Action(s) and Compliance Time(s):**

Required as indicated by this AD, unless the actions required by this AD have been already accomplished:

**Inspection(s):**

- (1) For Group 1 aeroplanes: Within 3 months after the effective date of this AD, accomplish a GVI of each affected part, except as specified in paragraph (2) of this AD, in accordance with the instructions of the AOT.
- (2) For aeroplanes that have Airbus modification (MOD) 161964 and MOD 161796 embodied in production, the inspection as required by paragraph (1) of this AD, does not need to be accomplished for affected parts installed on door 3 FIN position 7693MM and 7694MM.

**Corrective Action(s):**

- (3) If, during the inspection as required by paragraph (1) of this AD, discrepancies, as defined in the AOT, are detected on an affected part, before next flight, replace that affected part with a serviceable part, in accordance with the instructions of the AOT.
- (4) Replacing a reservoir with a reservoir not having an affected part installed is an acceptable method to comply with the requirement of paragraph (3) of this AD for that aeroplane, as applicable.

**Parts Installation:**

- (5) For Group 1 and Group 2 aeroplanes: From the effective date of this AD, do not install an affected part on any aeroplane.



**Ref. Publications:**

Airbus AOT 25N027-23 original issue dated 16 January 2024.

The use of later approved revisions of the above-mentioned document is acceptable for compliance with the requirements of this AD.

**Remarks:**

1. If requested and appropriately substantiated, EASA can approve Alternative Methods of Compliance for this AD.
2. This AD was posted on 31 January 2024 as PAD 24-013 for consultation until 14 February 2024. The Comment Response Document can be found in the [EASA Safety Publications Tool](#), in the compressed (zipped) file attached to the record for this AD.
3. Enquiries regarding this AD should be referred to the EASA Safety Information Section, Certification Directorate. E-mail: [ADs@easa.europa.eu](mailto:ADs@easa.europa.eu).
4. Information about any failures, malfunctions, defects or other occurrences, which may be similar to the unsafe condition addressed by this AD, and which may occur, or have occurred on a product, part or appliance not affected by this AD, can be reported to the [EU aviation safety reporting system](#). This may include reporting on the same or similar components, other than those covered by the design to which this AD applies, if the same unsafe condition can exist or may develop on an aircraft with those components installed. Such components may be installed under an FAA Parts Manufacturer Approval (PMA), Supplemental Type Certificate (STC) or other modification.
5. For any question concerning the technical content of the requirements in this AD, please contact: AIRBUS – Airworthiness Office – 1IASA; E-mail: [account.airworth-eas@airbus.com](mailto:account.airworth-eas@airbus.com).



Annex 1 – Affected Parts and Locations

	<b>P/N</b>	<b>Location</b>	<b>Functional Item Number (FIN)</b>
Escape slide inflation reservoir	70197-101	Left-hand (LH) emergency exit door 3 slide, overhead storage compartment (OHSC)	7693MM
	70197-101	Right-hand (RH) emergency exit door 3 slide, OHSC	7694MM
Offwing slide inflation reservoir	70200-101	LH offwing slide, cargo compartment	7683MM
	70200-102	RH offwing slide, cargo compartment	7684MM
	70200-103	LH offwing slide, OHSC	7685MM
	70200-104	RH offwing slide, OHSC	7686MM

