



## 適航指令發布單

## Airworthiness Directive Issuance Form

民航局 AD 編號 AD number	CAA-2023-08-004	發布日期 Date issued	2023/8/8
適用之航空產品 Applied to (models, serial numbers or part numbers, as applicable)	Airbus A330-201, A330-202, A330-203, A330-223, A330-223F, A330-243, A330-243F, A330-301, A330-302, A330-303, A330-321, A330-322, A330-323, A330-341, A330-342, A330-343, A330-841, A330-941 and A330-743L aeroplanes, all manufacturer serial numbers (MSN); and Airbus A340-211, A340-212, A340-213, A340-311, A340-312 and A340-313 aeroplanes, all MSN.		
主旨摘要 Subject	Airborne Auxiliary Power – Auxiliary Power Unit Scroll Housing Assembly – Replacement		
民航局 CAA <input type="checkbox"/> 本國產品 Native product <input type="checkbox"/> 其他個案 Other	設計國民航主管機構 Original Authority <input type="checkbox"/> FAA <input type="checkbox"/> Germany LBA <input checked="" type="checkbox"/> EASA <input type="checkbox"/> CAA-NL <input type="checkbox"/> Brazil <input type="checkbox"/> UK CAA <input type="checkbox"/> Transport Canada Civil Aviation <input type="checkbox"/> Japan CAB <input type="checkbox"/> DGAC <input type="checkbox"/> CAA of Israel <input type="checkbox"/> Other _____		
	設計國 AD 編號 Original AD number	2023-0158	
	1. 直接採用原 AD 之內容?(Is the original AD directly adopted?) <input checked="" type="checkbox"/> 是(Yes) <input type="checkbox"/> 否(No) _ a. 生效日期另訂為(Re-specify the effective date as) : _____ b. 執行時限另訂為(Re-specify the compliance time or period as) : _____ 2. 使用人是否需要將 AD 執行結果向民航局提出報告?(Do Users need to report the status of compliance to the CAA?) <input type="checkbox"/> 是(Yes) <input checked="" type="checkbox"/> 否(No)		
備註 Note	This AD supersedes EASA AD 2023-0056(CAA-2023-03-009) dated 16 March 2023.		
註： 1. AD 內容後附。 2. 航空器產品使用人得向民航局提出豁免、替代符合方法、執行時限之展延之申請。 3. 如有任何問題，請聯絡交通部民用航空局初始適航科。Tel：(02)2349-6330 / 6332, Fax：(02)2545-8464, e-mail： <a href="mailto:adcaa@mail.caa.gov.tw">adcaa@mail.caa.gov.tw</a> Note： 1. The AD text is enclosed. 2. Exemption, an alternative method of compliance or adjustment of the compliance time may be proposed to the CAA for approval. 3. For further information, please contact Civil Aeronautics Administration on Tel：(02)2349-6330 / 6332, Fax： (02)2545-8464, e-mail： <a href="mailto:adcaa@mail.caa.gov.tw">adcaa@mail.caa.gov.tw</a>			



## Airworthiness Directive

**AD No.:** 2023-0158

**Issued:** 02 August 2023

Note: This Airworthiness Directive (AD) is issued by EASA, acting in accordance with Regulation (EU) 2018/1139 on behalf of the European Union, its Member States and of the European third countries that participate in the activities of EASA under Article 129 of that Regulation.

This AD is issued in accordance with Regulation (EU) 748/2012, Part 21.A.3B. In accordance with Regulation (EU) 1321/2014 Annex I Part M.A.301, or Annex Vb Part ML.A.301, as applicable, the continuing airworthiness of an aircraft shall be ensured by accomplishing any applicable ADs. Consequently, no person may operate an aircraft to which an AD applies, except in accordance with the requirements of that AD, unless otherwise specified by the Agency [Regulation (EU) 1321/2014 Annex I Part M.A.303, or Annex Vb Part ML.A.303, as applicable] or agreed with the Authority of the State of Registry [Regulation (EU) 2018/1139, Article 71 exemption].

### Design Approval Holder's Name:

AIRBUS S.A.S.

### Type/Model designation(s):

A330 and A340 aeroplanes

**Effective Date:** 16 August 2023

**TCDS Numbers:** EASA.A.004, EASA.A.015

**Foreign AD:** Not applicable

**Supersedure:** This AD supersedes EASA AD 2023-0056 dated 16 March 2023.

## ATA 49 – Airborne Auxiliary Power – Auxiliary Power Unit Scroll Housing Assembly – Replacement

### Manufacturer(s):

Airbus, formerly Airbus Industrie

### Applicability:

Airbus A330-201, A330-202, A330-203, A330-223, A330-223F, A330-243, A330-243F, A330-301, A330-302, A330-303, A330-321, A330-322, A330-323, A330-341, A330-342, A330-343, A330-841, A330-941 and A330-743L aeroplanes, all manufacturer serial numbers (MSN); and

Airbus A340-211, A340-212, A340-213, A340-311, A340-312 and A340-313 aeroplanes, all MSN.

### Definitions:

For the purpose of this AD, the following definitions apply:

**The SB:** Honeywell Service Bulletin (SB) 5053-181-49-7895.

**Affected part:** Auxiliary Power Unit (APU) Scroll Housing Assemblies, having Part Number (P/N) 5053-181-001-501 or P/N 0331207990.

**SR-1 affected part:** An affected part marked with 'SR-1' next to its P/N marking.



**Serviceable part:** APU Scroll Housing Assemblies, having P/N 5053-181-002-501 or P/N 5053-881-002-501.

**Affected APU:** Honeywell GTCP331-350[C] APUs, all serial numbers up to ( )904 inclusive, that have an affected part installed. For the purpose of this AD, any letter included in the APU s/n shall be disregarded.

**Serviceable APU:** Any Honeywell GTCP331-350[C] APU which is not an affected APU. The SB provides instructions to modify an affected APU into a serviceable APU.

**Groups:** Group 1 aeroplanes are those that have an affected part installed. Group 2 aeroplanes are those that do not have an affected part installed.

**Reason:**

Occurrences have been reported where cracks were found in the scroll housing assembly of certain Honeywell GTCP331-350[C] APUs.

This condition, if not corrected, could lead to hot air leakage, with consequent damage to the APU compartment and loss of the APU doors, possibly resulting in damage to the aeroplane.

To address this potential unsafe condition, Honeywell issued the SB, identifying the affected APU and providing instructions to replace the affected part with a serviceable part, as defined in this AD. Consequently, EASA issued AD 2023-0056 to require replacement of each affected APU with a serviceable APU, and to prohibit (re)installation of affected APU.

Since that AD was issued, it has been determined that a letter, which is part of an APU's serial number, is only used to identify the manufacturing facility of the APU, and has no meaning in relation to the AD required actions.

For the reason described above, this AD retains the requirements of EASA AD 2023-0056, which is superseded, but provides an updated definition of 'affected APU'.

**Required Action(s) and Compliance Time(s):**

Required as indicated, unless accomplished previously:

**Modification:**

- (1) For Group 1 aeroplanes: Within 36 months after 30 March 2023 [the effective date of EASA AD 2023-0056], replace the affected APU with a serviceable APU. This can be accomplished in accordance with the instructions of the applicable Airbus aircraft maintenance manual (AMM).
- (2) Re-identification of an SR-1 affected part on an aeroplane in accordance with the instructions of the SB is an acceptable method to comply with the requirements of paragraph (1) of this AD for that aeroplane.

**Part Installation:**

- (3) Do not install an affected APU on any aeroplane, as required by paragraph (3.1) or (3.2) of this AD, as applicable.



(3.1) For Group 1 aeroplanes: After modification of the aeroplane as required by paragraph (1) of this AD, or as specified in paragraph (2) of this AD, as applicable.

(3.2) For Group 2 aeroplanes: From the effective date of this AD.

**Ref. Publications:**

Honeywell SB 5053-181-49-7895 original issue dated 21 July 2006.

**Remarks:**

1. If requested and appropriately substantiated, EASA can approve Alternative Methods of Compliance for this AD.
2. Based on the required actions and the compliance time, EASA have decided to issue a Final AD with Request for Comments, postponing the public consultation process until after publication.
3. Enquiries regarding this AD should be referred to the EASA Safety Information Section, Certification Directorate. E-mail: [ADs@easa.europa.eu](mailto:ADs@easa.europa.eu).
4. Information about any failures, malfunctions, defects or other occurrences, which may be similar to the unsafe condition addressed by this AD, and which may occur, or have occurred on a product, part or appliance not affected by this AD, can be reported to the [EU aviation safety reporting system](#). This may include reporting on the same or similar components, other than those covered by the design to which this AD applies, if the same unsafe condition can exist or may develop on an aircraft with those components installed. Such components may be installed under an FAA Parts Manufacturer Approval (PMA), Supplemental Type Certificate (STC) or other modification.
5. For any question concerning the technical content of the requirements in this AD, please contact: AIRBUS – 1IAL (Airworthiness Office), E-mail: [airworthiness.A330-A340@airbus.com](mailto:airworthiness.A330-A340@airbus.com).

