



適航指令發布單
Airworthiness Directive Issuance Form

民航局 AD 編號 AD number	CAA-2022-05-003	發布日期 Date issued	2022/5/16												
適用之航空產品 Applied to (models, serial numbers or part numbers, as applicable)	Airbus A350-941 and A350-1041 aeroplanes, manufacturer serial numbers 0006 through 0058 (inclusive), 0060 through 0107, 0109 through 0130, 0132 through 0138, 0140 through 0151, 0153 through 0155, 0157 through 0165, 0167 through 0170, 0172 through 0193, 0195 through 0221, 0223 through 0241, 0243 through 0252, 0254 through 0269, 0271 through 0274, 0276 through 0282, 0284 through 0287, 0289 through 0303, 0305 through 0316, 0318 through 0352, 0354 through 0359, 0362 through 0366, 0368 through 0376, 0378 through 0383, 0385 through 0408, 0410 through 0429, 0431 through 0437, 0439, 0441 through 0443, 0445 through 0449, 0451, 0452, 0454, 0455, 0458, 0464, 0465, 0468 through 0470, 0472, 0474 through 0476, 0479, 0481, 0482, 0484, 0488, 0489, 0491, 0497, 0499, 0500, 0502, 0517, 0531, 0535 and 0536.														
主旨摘要 Subject	Wings - Centre Wing Box Fasteners - Replacement														
民航局 CAA <input type="checkbox"/> 本國產品 Native product <input type="checkbox"/> 其他個案 Other	設計國民航主管機構 Original Authority <table border="0" style="width: 100%;"><tr><td><input type="checkbox"/> FAA</td><td><input type="checkbox"/> Germany LBA</td></tr><tr><td><input checked="" type="checkbox"/> EASA</td><td><input type="checkbox"/> CAA-NL</td></tr><tr><td><input type="checkbox"/> Brazil</td><td><input type="checkbox"/> UK CAA</td></tr><tr><td><input type="checkbox"/> Transport Canada Civil Aviation</td><td><input type="checkbox"/> Japan CAB</td></tr><tr><td><input type="checkbox"/> DGAC</td><td><input type="checkbox"/> CAA of Israel</td></tr><tr><td></td><td><input type="checkbox"/> Other _____</td></tr></table>			<input type="checkbox"/> FAA	<input type="checkbox"/> Germany LBA	<input checked="" type="checkbox"/> EASA	<input type="checkbox"/> CAA-NL	<input type="checkbox"/> Brazil	<input type="checkbox"/> UK CAA	<input type="checkbox"/> Transport Canada Civil Aviation	<input type="checkbox"/> Japan CAB	<input type="checkbox"/> DGAC	<input type="checkbox"/> CAA of Israel		<input type="checkbox"/> Other _____
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<input type="checkbox"/> Transport Canada Civil Aviation	<input type="checkbox"/> Japan CAB														
<input type="checkbox"/> DGAC	<input type="checkbox"/> CAA of Israel														
	<input type="checkbox"/> Other _____														
	設計國 AD 編號 Original AD number	2022-0080													
	1. 直接採用原 AD 之內容?(Is the original AD directly adopted?) <input checked="" type="checkbox"/> 是(Yes) <input type="checkbox"/> 否(No) _ a. 生效日期另訂為(Re-specify the effective date as) : _____ b. 執行時限另訂為(Re-specify the compliance time or period as) : _____ 2. 使用人是否需要將 AD 執行結果向民航局提出報告?(Do Users need to report the status of compliance to the CAA?) <input type="checkbox"/> 是(Yes) <input checked="" type="checkbox"/> 否(No)														
備註 Note	ATA 57. Ref. Publications: Airbus SB A350-57-P069 original issue dated 16 November 2021.														

- 註：
1. AD 內容後附。
 2. 航空器產品使用人得向民航局提出豁免、替代符合方法、執行時限之展延之申請。
 3. 如有任何問題，請聯絡交通部民用航空局初始適航科。Tel：(02)2349-6330 / 6332, Fax：(02)2545-8464, e-mail：adcaa@mail.caa.gov.tw
- Note：
1. The AD text is enclosed.
 2. Exemption, an alternative method of compliance or adjustment of the compliance time may be proposed to the CAA for approval.
 3. For further information, please contact Civil Aeronautics Administration on Tel：(02)2349-6330 / 6332, Fax：(02)2545-8464, e-mail：adcaa@mail.caa.gov.tw

CAA Form ACS-P08-02

第一頁/共一頁



Airworthiness Directive

AD No.: 2022-0080

Issued: 09 May 2022

Note: This Airworthiness Directive (AD) is issued by EASA, acting in accordance with Regulation (EU) 2018/1139 on behalf of the European Union, its Member States and of the European third countries that participate in the activities of EASA under Article 129 of that Regulation.

This AD is issued in accordance with Regulation (EU) 748/2012, Part 21.A.3B. In accordance with Regulation (EU) 1321/2014 Annex I, Part M.A.301 or Annex Vb Part ML.A.301, as applicable, the continuing airworthiness of an aircraft shall be ensured by accomplishing any applicable ADs. Consequently, no person may operate an aircraft to which an AD applies, except in accordance with the requirements of that AD, unless otherwise specified by the Agency [Regulation (EU) 1321/2014 Annex I, Part M.A.303 or Annex Vb Part ML.A.303, as applicable] or agreed with the Authority of the State of Registry [Regulation (EU) 2018/1139, Article 71 exemption].

Design Approval Holder's Name:

AIRBUS S.A.S.

Type/Model designation(s):

A350 aeroplanes

Effective Date: 23 May 2022

TCDS Number(s): EASA.A.151

Foreign AD: Not applicable

Supersedure: None

ATA 57 – Wings – Centre Wing Box Fasteners – Replacement

Manufacturer(s):

Airbus

Applicability:

Airbus A350-941 and A350-1041 aeroplanes, manufacturer serial numbers 0006 through 0058 (inclusive), 0060 through 0107, 0109 through 0130, 0132 through 0138, 0140 through 0151, 0153 through 0155, 0157 through 0165, 0167 through 0170, 0172 through 0193, 0195 through 0221, 0223 through 0241, 0243 through 0252, 0254 through 0269, 0271 through 0274, 0276 through 0282, 0284 through 0287, 0289 through 0303, 0305 through 0316, 0318 through 0352, 0354 through 0359, 0362 through 0366, 0368 through 0376, 0378 through 0383, 0385 through 0408, 0410 through 0429, 0431 through 0437, 0439, 0441 through 0443, 0445 through 0449, 0451, 0452, 0454, 0455, 0458, 0464, 0465, 0468 through 0470, 0472, 0474 through 0476, 0479, 0481, 0482, 0484, 0488, 0489, 0491, 0497, 0499, 0500, 0502, 0517, 0531, 0535 and 0536.

Definitions:

For the purpose of this AD, the following definitions apply:

The SB: Airbus Service Bulletin (SB) A350-57-P069.

Affected parts: Fasteners, installed on left-hand and right-hand centre wing box (CWB) at front and rear spar area, as specified in the SB.



Reason:

During flight and fatigue test, it was detected that some fasteners rotate inside their fastener holes of the CWB. Further investigation identified insufficient friction for the application.

This condition, if not corrected, could lead to a loss of fastener clamping and crack of nut sealant cover, possibly resulting, in case of lightning strike, in a fuel tank explosion and consequent loss of the aeroplane.

To address this potential unsafe condition, Airbus issued the SB to provide instructions to replace the affected parts and to introduce additional head nut cap protection at front and rear spar in the CWB.

For the reasons described above, this AD requires replacement of the affected parts and introduction of additional head nut cap protection at front and rear spar in the CWB.

Required Action(s) and Compliance Time(s):

Required as indicated, unless accomplished previously:

Modification:

Within 69 months after the effective date of this AD, replace the affected parts and apply additional head nut cap protection at front and rear spar in the CWB in accordance with the instructions of the SB.

Ref. Publications:

Airbus SB A350-57-P069 original issue dated 16 November 2021.

The use of later approved revisions of the above-mentioned document is acceptable for compliance with the requirements of this AD.

Remarks:

1. If requested and appropriately substantiated, EASA can approve Alternative Methods of Compliance for this AD.
2. This AD was posted on 22 March 2022 as PAD 22-029 for consultation until 19 April 2022. The Comment Response Documents can be found in the [EASA Safety Publications Tool](#), in the compressed (zipped) file attached to the record for this AD.
3. Enquiries regarding this AD should be referred to the EASA Safety Information Section, Certification Directorate. E-mail: ADs@easa.europa.eu.
4. Information about any failures, malfunctions, defects or other occurrences, which may be similar to the unsafe condition addressed by this AD, and which may occur, or have occurred on a product, part or appliance not affected by this AD, can be reported to the [EU aviation safety reporting system](#). This may include reporting on the same or similar components, other than those covered by the design to which this AD applies, if the same unsafe condition can exist or may develop on an aircraft with those components installed. Such components may be



installed under an FAA Parts Manufacturer Approval (PMA), Supplemental Type Certificate (STC) or other modification.

5. For any question concerning the technical content of the requirements in this AD, please contact: AIRBUS A350 XWB, E-mail: continued-airworthiness.a350@airbus.com.

