



**適航指令發布單**  
**Airworthiness Directive Issuance Form**

民航局 AD 編號 AD number	CAA-2021-08-010	發布日期 Date issued	2021/9/2												
適用之航空產品 Applied to (models, serial numbers or part numbers, as applicable)	AW169 helicopters, all serial numbers (s/n).														
主旨摘要	Rotors Flight Control - Yaw Pedal Assemblies - Modification														
民航局 CAA <input type="checkbox"/> 本國產品 Native products  <input type="checkbox"/> 其他個案 Other	設計國民航主管機構 Original Authorities <table border="0" style="width: 100%;"><tr><td><input type="checkbox"/> FAA</td><td><input type="checkbox"/> Germany LBA</td></tr><tr><td><input checked="" type="checkbox"/> EASA</td><td><input type="checkbox"/> CAA-NL</td></tr><tr><td><input type="checkbox"/> Brazil</td><td><input type="checkbox"/> UK CAA</td></tr><tr><td><input type="checkbox"/> Transport Canada Civil Aviation</td><td><input type="checkbox"/> Japan CAB</td></tr><tr><td><input type="checkbox"/> DGAC</td><td><input type="checkbox"/> CAA of Israel</td></tr><tr><td></td><td><input type="checkbox"/> Other _____</td></tr></table>			<input type="checkbox"/> FAA	<input type="checkbox"/> Germany LBA	<input checked="" type="checkbox"/> EASA	<input type="checkbox"/> CAA-NL	<input type="checkbox"/> Brazil	<input type="checkbox"/> UK CAA	<input type="checkbox"/> Transport Canada Civil Aviation	<input type="checkbox"/> Japan CAB	<input type="checkbox"/> DGAC	<input type="checkbox"/> CAA of Israel		<input type="checkbox"/> Other _____
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	<input type="checkbox"/> Other _____														
	設計國 AD 編號 Original AD number	2021-0199													
	1. 直接採用原 AD 之內容?(Is the original AD directly adopted?) <input checked="" type="checkbox"/> 是(Yes) <input type="checkbox"/> 否(No) _ a. 生效日期另訂為(Re-specify the effective date as) : _____ b. 執行時限另訂為(Re-specify the compliance time or period as) : _____ <input type="checkbox"/> 2. 使用人是否需要將 AD 執行結果向民航局提出報告?(Do <input type="checkbox"/> Users need to report the status of compliance to the CAA?) <input type="checkbox"/> 是(Yes) <input checked="" type="checkbox"/> 否(No)														
備註 Note	This AD supersedes EASA AD 2019-0252(CAA-2019-10-009) issued on 10 October 2019.														
<p>註： 1. AD 內容後附。 2. 航空器產品使用人得向民航局提出豁免、替代符合方法、執行時限之展延之申請。 3. 如有任何問題，請聯絡交通部民用航空局初始適航科。Tel：(02)2349-6331~3, Fax：(02)2545-8464, e-mail：<a href="mailto:adcaa@mail.caa.gov.tw">adcaa@mail.caa.gov.tw</a></p> <p>Note： 1. The AD text is enclosed. 2. Exemption, an alternative method of compliance or adjustment of the compliance time may be proposed to the CAA for approval. 3. For further information, please contact Civil Aeronautics Administration on Tel：(02)2349-6331~3, Fax：(02)2545-8464, e-mail：<a href="mailto:adcaa@mail.caa.gov.tw">adcaa@mail.caa.gov.tw</a></p>															



## Airworthiness Directive

**AD No.:** 2021-0199

**Issued:** 27 August 2021

Note: This Airworthiness Directive (AD) is issued by EASA, acting in accordance with Regulation (EU) 2018/1139 on behalf of the European Union, its Member States and of the European third countries that participate in the activities of EASA under Article 129 of that Regulation.

This AD is issued in accordance with Regulation (EU) 748/2012, Part 21.A.3B. In accordance with Regulation (EU) 1321/2014 Annex I, Part M.A.301, the continuing airworthiness of an aircraft shall be ensured by accomplishing any applicable ADs. Consequently, no person may operate an aircraft to which an AD applies, except in accordance with the requirements of that AD, unless otherwise specified by the Agency [Regulation (EU) 1321/2014 Annex I, Part M.A.303] or agreed with the Authority of the State of Registry [Regulation (EU) 2018/1139, Article 71 exemption].

**Design Approval Holder's Name:**

LEONARDO S.p.A.

**Type/Model designation(s):**

AW169 helicopters

**Effective Date:** 10 September 2021

**TCDS Number(s):** EASA.R.509

**Foreign AD:** Not applicable

**Supersedure:** This AD supersedes EASA AD 2019-0252 issued on 10 October 2019.

### ATA 67 – Rotors Flight Control – Yaw Pedal Assemblies – Modification

**Manufacturer(s):**

Leonardo S.p.A. Helicopters, formerly Finmeccanica S.p.A., AgustaWestland S.p.A.

**Applicability:**

AW169 helicopters, all serial numbers (s/n).

**Definitions:**

For the purpose of this AD, the following definitions apply:

**Affected part:** Pilot yaw pedal assemblies, having Part Number (P/N) 6F6720A06831 or P/N 6F6720A06832, including those marked with retromod 6F6720P01231; and co-pilot yaw pedal assemblies, having P/N 6F6720A06931 or P/N 6F6720A06932, including those marked with retromod P/N 6F6720P01331.

**Non-affected part:** Pilot yaw pedal assemblies, having P/N 6F6720A06831 or P/N 6F6720A06832 and marked with retromod P/N 6F6720P01731, or pilot yaw pedal assembly P/N 6F6720A06833; and co-pilot yaw pedal assemblies, having P/N 6F6720A06931 or P/N 6F6720A06932 and marked with retromod P/N 6F6720P01831, or co-pilot yaw pedal assembly P/N 6F6720A06933.

**The SB:** Leonardo Service Bulletin (SB) 169-181.



**Groups:** Group 1 helicopters are those that have an affected part installed.

Group 2 helicopters are those that do not have an affected part installed. A helicopter on which Leonardo modification (mod) 6F6720P01411 has been embodied in production is Group 2, provided the helicopter remains in that configuration.

**Reason:**

An occurrence was reported of finding a broken adjustable device, part of the pilot and co-pilot yaw pedal assembly. The results of the investigations determined that a modification of the pedal assembly was required to prevent the effect of this kind of failure.

This condition, if not corrected, could possibly result in reduced yaw control of the helicopter.

To initially address this unsafe condition, Leonardo developed a modification/rework of the pilot and co-pilot pedal assembly and published SB 169-147 to provide modification instructions. Consequently, EASA issued AD 2019-0252 to require rework of the pilot and co-pilot pedal assemblies. That AD was considered an interim action.

Since that AD was issued, three additional events have been reported where the universal joint of the adjusting mechanism on the yaw pedals failed.

Prompted by these findings, Leonardo developed mod 6F6720P01411, introducing upgraded pilot and co-pilot pedal assemblies of improved design which removes the failure modes. Leonardo also issued the SB to provide in-service modification instructions.

For the reasons described above, this AD cancels the requirements of EASA AD 2019-0252, which is superseded, and requires modification and re-identification of the affected parts.

**Required Action(s) and Compliance Time(s):**

Required as indicated, unless accomplished previously:

**Modification:**

- (1) For Group 1 helicopters: Within 400 flight hours or 12 months, whichever occurs first after the effective date of this AD, modify each installed affected part, as defined in this AD, and re-identify each modified pilot and co-pilot pedal assembly in accordance with the instructions of the SB.
- (2) Replacing each installed affected part on a helicopter with a non-affected part in accordance with the instructions of the SB is an acceptable method to comply with the modification requirements of paragraph (1) of this AD for that helicopter.

**Parts Installation:**

- (3) Do not install on any helicopter an affected part, as defined in this AD, as required by paragraph (3.1) or (3.2) of this AD, as applicable.
  - (3.1) For Group 1 helicopters: After modification of the helicopter as required by paragraph (1) of this AD.



(3.2) For Group 2 helicopters: From the effective date of this AD.

**Ref. Publications:**

Leonardo S.p.A. SB 169-181 original issue dated 26 July 2021.

The use of later approved revisions of the above-mentioned document is acceptable for compliance with the requirements of this AD.

**Remarks:**

1. If requested and appropriately substantiated, EASA can approve Alternative Methods of Compliance for this AD.
2. This AD was posted on 28 July 2021 as PAD 21-110 for consultation until 25 August 2021. No comments were received during the consultation period.
3. Enquiries regarding this AD should be referred to the EASA Safety Information Section, Certification Directorate. E-mail: [ADs@easa.europa.eu](mailto:ADs@easa.europa.eu).
4. Information about any failures, malfunctions, defects or other occurrences, which may be similar to the unsafe condition addressed by this AD, and which may occur, or have occurred on a product, part or appliance not affected by this AD, can be reported to the [EU aviation safety reporting system](#). This may include reporting on the same or similar components, other than those covered by the design to which this AD applies, if the same unsafe condition can exist or may develop on an aircraft with those components installed. Such components may be installed under an FAA Parts Manufacturer Approval (PMA), Supplemental Type Certificate (STC) or other modification.
5. For any question concerning the technical content of the requirements in this AD, please contact: Leonardo S.p.A. Helicopters, E-mail: [PSE\\_AW169.MBX.AW@leonardocompany.com](mailto:PSE_AW169.MBX.AW@leonardocompany.com).

