



適航指令發布單

Airworthiness Directive Issuance Form

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|---|--|---------------------|----------|
| 民航局 AD 編號 AD number | CAA-2021-08-001 | 發布日期 Date issued | 2021/8/9 |
| 適用之航空產品 Applied to (models, serial numbers or part numbers, as applicable) | Airbus A318-111, A318-112, A318-121, A318-122, A319-111, A319-112, A319-113, A319-114, A319-115, A319-131, A319-132, A319-133, A319-151N, A319-153N, A319-171N, A320-211, A320-212, A320-214, A320-215, A320-216, A320-231, A320-232, A320-233, A320-251N, A320-252N, A320-253N, A320-271N, A320-272N, A320-273N, A321-111, A321-112, A321-131, A321-211, A321-212, A321-213, A321-231, A321-232, A321-251N, A321-251NX, A321-252N, A321-252NX, A321-253N, A321-253NX, A321-271N, A321-271NX, A321-272N and A321-272NX aeroplanes, all manufacturer serial numbers. | | |
| 主旨摘要 | Equipment / Furnishings - Galleys - Inspection | | |
| 民航局 CAA <input type="checkbox"/> 本國產品 Native products <input type="checkbox"/> 其他個案 Other | 設計國民航主管機構 Original Authorities <div style="display: flex; justify-content: space-between;"> <div style="width: 48%;"> <input type="checkbox"/> FAA <input checked="" type="checkbox"/> EASA <input type="checkbox"/> Brazil <input type="checkbox"/> Transport Canada Civil Aviation <input type="checkbox"/> DGAC </div> <div style="width: 48%;"> <input type="checkbox"/> Germany LBA <input type="checkbox"/> CAA-NL <input type="checkbox"/> UK CAA <input type="checkbox"/> Japan CAB <input type="checkbox"/> CAA of Israel <input type="checkbox"/> Other _____ </div> </div> | | |
| | 設計國 AD 編號 Original AD number | 2021-0183 | |
| | 1. 直接採用原 AD 之內容?(Is the original AD directly adopted?) <input checked="" type="checkbox"/> 是(Yes) <input type="checkbox"/> 否(No) _ a. 生效日期另訂為(Re-specify the effective date as) : _____ b. 執行時限另訂為(Re-specify the compliance time or period as) : _____ <input type="checkbox"/> | | |
| | 2. 使用人是否需要將 AD 執行結果向民航局提出報告?(Do <input type="checkbox"/> Users need to report the status of compliance to the CAA?) <input type="checkbox"/> 是(Yes) <input checked="" type="checkbox"/> 否(No) | | |
| 備註 Note | ATA 25. Ref. Publications: Airbus SB A320-25-1BVS original issue dated 18 December 2020. and Airbus SB A320-25-1BVT original issue dated 18 December 2020. | | |
| 註： 1. AD 內容後附。 2. 航空器產品使用人得向民航局提出豁免、替代符合方法、執行時限之展延之申請。 3. 如有任何問題，請聯絡交通部民用航空局初始適航科。Tel：(02)2349-6331~3, Fax：(02)2545-8464, e-mail： adcaa@mail.caa.gov.tw Note： 1. The AD text is enclosed. 2. Exemption, an alternative method of compliance or adjustment of the compliance time may be proposed to the CAA for approval. 3. For further information, please contact Civil Aeronautics Administration on Tel：(02)2349-6331~3, Fax：(02)2545-8464, e-mail： adcaa@mail.caa.gov.tw | | | |



Airworthiness Directive

AD No.: 2021-0183

Issued: 04 August 2021

Note: This Airworthiness Directive (AD) is issued by EASA, acting in accordance with Regulation (EU) 2018/1139 on behalf of the European Union, its Member States and of the European third countries that participate in the activities of EASA under Article 129 of that Regulation.

This AD is issued in accordance with Regulation (EU) 748/2012, Part 21.A.3B. In accordance with Regulation (EU) 1321/2014 Annex I, Part M.A.301, the continuing airworthiness of an aircraft shall be ensured by accomplishing any applicable ADs. Consequently, no person may operate an aircraft to which an AD applies, except in accordance with the requirements of that AD, unless otherwise specified by the Agency [Regulation (EU) 1321/2014 Annex I, Part M.A.303] or agreed with the Authority of the State of Registry [Regulation (EU) 2018/1139, Article 71 exemption].

Design Approval Holder's Name:

AIRBUS

Type/Model designation(s):

A318, A319, A320 and A321 aeroplanes

Effective Date: 18 August 2021

TCDS Number(s): EASA.A.064

Foreign AD: Not applicable

Supersedure: None

ATA 25 – Equipment / Furnishings – Galleys – Inspection

Manufacturer(s):

Airbus, formerly Airbus Industrie

Applicability:

Airbus A318-111, A318-112, A318-121, A318-122, A319-111, A319-112, A319-113, A319-114, A319-115, A319-131, A319-132, A319-133, A319-151N, A319-153N, A319-171N, A320-211, A320-212, A320-214, A320-215, A320-216, A320-231, A320-232, A320-233, A320-251N, A320-252N, A320-253N, A320-271N, A320-272N, A320-273N, A321-111, A321-112, A321-131, A321-211, A321-212, A321-213, A321-231, A321-232, A321-251N, A321-251NX, A321-252N, A321-252NX, A321-253N, A321-253NX, A321-271N, A321-271NX, A321-272N and A321-272NX aeroplanes, all manufacturer serial numbers.

Definitions:

For the purpose of this AD, the following definitions apply:

Affected part: Forward-facing galleys, having a Part Number (P/N) as listed in Appendix 1 of this AD.

The applicable SB: Airbus Service Bulletin (SB) A320-25-1BVS or Airbus SB A320-25-1BVT, as applicable.

Aeroplane date of manufacture: The date of transfer of title (ownership) at the time of first delivery to an operator, which is referenced in Airbus documentation.



Groups: Group 1 aeroplanes are those that have an affected part installed. Group 2 aeroplanes are those that do not have an affected part installed.

Reason:

Occurrences have been reported of finding damage during inspection of certain galleys, including delamination of work deck as well as corroded and cracked retainer blocks.

This condition, if not detected and corrected, could affect the galley's capability to hold the trolley under emergency landing loads, which could lead to trolley detachment, possibly resulting in blocking of an escape path during an emergency exit.

To address this potential unsafe condition, Airbus issued the applicable SB, providing instructions to inspect the affected parts.

For the reasons described above, this AD requires repetitive general visual inspection (GVI) of each affected part and, depending on findings, accomplishment of applicable corrective action(s).

This AD is considered to be an interim action and further AD action may follow.

Required Action(s) and Compliance Time(s):

Required as indicated, unless accomplished previously:

Inspection(s):

- (1) For Group 1 aeroplanes: Within 36 months since first installation of an affected part on an aeroplane, or within 12 months after the effective date of this AD, whichever occurs later, and, thereafter, at intervals not exceeding 6 months, accomplish a GVI of each affected part in accordance with the instructions of the applicable SB (see Note 1 of this AD).

Note 1: If the date of first installation of an affected part on an aeroplane is unknown, the aeroplane date of manufacture can be used instead.

Corrective Action(s):

- (2) If, during any GVI as required by paragraph (1) of this AD, discrepancies are detected on an affected part, before next flight, accomplish the applicable corrective action(s) in accordance with the instructions of the applicable SB.

Terminating Action:

- (3) None.

Part(s) Installation:

- (4) For Group 1 and Group 2 aeroplanes: From the effective date of this AD, it is allowed to install on any aeroplane an affected part, provided that, following installation, it is inspected and, depending on findings, corrected, as required by this AD.

Ref. Publications:

Airbus SB A320-25-1BVS original issue dated 18 December 2020.



Airbus SB A320-25-1BVT original issue dated 18 December 2020.

The use of later approved revisions of the above-mentioned documents is acceptable for compliance with the requirements of this AD.

Remarks:

1. If requested and appropriately substantiated, EASA can approve Alternative Methods of Compliance for this AD.
2. This AD was posted on 29 January 2021 as PAD 21-005 for consultation until 26 February 2021. The Comment Response Document can be found in the [EASA Safety Publications Tool](#), in the compressed (zipped) file attached to the record for this AD.
3. Enquiries regarding this AD should be referred to the EASA Safety Information Section, Certification Directorate. E-mail: ADs@easa.europa.eu.
4. Information about any failures, malfunctions, defects or other occurrences, which may be similar to the unsafe condition addressed by this AD, and which may occur, or have occurred on a product, part or appliance not affected by this AD, can be reported to the [EU aviation safety reporting system](#). This may include reporting on the same or similar components, other than those covered by the design to which this AD applies, if the same unsafe condition can exist or may develop on an aircraft with those components installed. Such components may be installed under an FAA Parts Manufacturer Approval (PMA), Supplemental Type Certificate (STC) or other modification.
5. For any question concerning the technical content of the requirements in this AD, please contact: AIRBUS – Airworthiness Office – IIASA; E-mail: account.airworth-eas@airbus.com.



Appendix 1 – Affected Parts (see Note 2 of this AD)

| Unit | Affected P/N |
|--------------------|---|
| Galley G2a | 601865-XXXXXX, 601866-XXXXXX, 611867-XXXXXX, 601897-XXXXXX, 601924-XXXXXX, 601937-XXXXXX, 601951-XXXXXX, 601957-XXXXXX, 601963-XXXXXX, 6019A8-XXXXXX, 6049C7-XXXXXX and 6019D6-XXXXXX |
| Galley G4x and SG4 | 601853-XXXXXX, 601854-XXXXXX, 601855-XXXXXX, 601557-XXXXXX, 601883-XXXXXX, 601888-XXXXXX, 601889-XXXXXX, 601891-XXXXXX, 6018A4-XXXXXX, 601930-XXXXXX, 601940-XXXXXX, 6019A3-XXXXXX, 601903-XXXXXX and 601958-XXXXXX |
| Galley G5 | 601856-XXXXXX, 601857-XXXXXX, 601858-XXXXXX, 601920-XXXXXX, 601983-XXXXXX, 6019A9-XXXXXX, 6019C1-XXXXXX and 6019F1-XXXXXX |
| Galley G8 | 601885-XXXXXX and 601935-XXXXXX |
| Aft Complex | 601537-XXXXXX, 601539-XXXXXX and 601567-XXXXXX |

Note 2: 'XXXXXX' represents any numerical combination of 1 or more (up to 6) digits.

