



## 適航指令發布單

### Airworthiness Directive Issuance Form

民航局 AD 編號 AD number	CAA-2021-02-013	發布日期 Date issued	2021/2/25
適用之航空產品 Applied to (models, serial numbers or part numbers, as applicable)	models EMB-145, EMB-145ER, EMB-145EU, EMB-145EP, EMB-145LR, EMB-145MR, EMB-145LU, EMB-145MP, EMB-145MK, EMB-145XR, EMB-135BJ, EMB-135ER, EMB-135KE, EMB-135KL and EMB-135LR, all serial numbers.		
主旨摘要	This AD requires inspection, and replacement of the MLG Lower Side Arm, inspection and reinstallation of the MLG Side Arm Strut, and correct installation of the Lower Side Arm Bearings.		
民航局 CAA <input type="checkbox"/> 本國產品 Native products <input type="checkbox"/> 其他個案 Other	設計國民航主管機構 Original Authorities <div style="display: flex; justify-content: space-between;"> <div style="width: 48%;"> <input type="checkbox"/> FAA  <input type="checkbox"/> EASA  <input checked="" type="checkbox"/> Brazil  <input type="checkbox"/> Transport Canada Civil Aviation  <input type="checkbox"/> DGAC             </div> <div style="width: 48%;"> <input type="checkbox"/> Germany LBA  <input type="checkbox"/> CAA-NL  <input type="checkbox"/> UK CAA  <input type="checkbox"/> Japan CAB  <input type="checkbox"/> CAA of Israel  <input type="checkbox"/> Other _____             </div> </div>		
	設計國 AD 編號 Original AD number	2021-02-02	
	1. 直接採用原 AD 之內容?(Is the original AD directly adopted?) <input checked="" type="checkbox"/> 是(Yes) <input type="checkbox"/> 否(No) _ a. 生效日期另訂為(Re-specify the effective date as) : _____ b. 執行時限另訂為(Re-specify the compliance time or period as) : _____ <input type="checkbox"/>		
	2. 使用人是否需要將 AD 執行結果向民航局提出報告?(Do <input type="checkbox"/> Users need to report the status of compliance to the CAA?) <input type="checkbox"/> 是(Yes) <input checked="" type="checkbox"/> 否(No)		
備註 Note	ATA 32. Ref. Embraer Maintenance Practices and Procedures, MPP1124_32-10-03-04-1, Revision 0, dated March 09, 2020, or later approved revisions.		
註： 1. AD 內容後附。 2. 航空器產品使用人得向民航局提出豁免、替代符合方法、執行時限之展延之申請。 3. 如有任何問題，請聯絡交通部民用航空局初始適航科。Tel：(02)2349-6331~3, Fax：(02)2545-8464, e-mail： <a href="mailto:adcaa@mail.caa.gov.tw">adcaa@mail.caa.gov.tw</a> Note： 1. The AD text is enclosed. 2. Exemption, an alternative method of compliance or adjustment of the compliance time may be proposed to the CAA for approval. 3. For further information, please contact Civil Aeronautics Administration on Tel：(02)2349-6331~3, Fax：(02)2545-8464, e-mail： <a href="mailto:adcaa@mail.caa.gov.tw">adcaa@mail.caa.gov.tw</a>			



**AGÊNCIA NACIONAL DE AVIAÇÃO CIVIL - BRAZIL**

## **BRAZILIAN AIRWORTHINESS DIRECTIVE**

**AD No.: 2021-02-02**

**Effective Date: 26 Feb. 2021**

The following Brazilian Airworthiness Directive (AD), issued by the Agência Nacional de Aviação Civil (ANAC) in accordance with provisions of Chapter IV, Title III of Código Brasileiro de Aeronáutica - Law No. 7,565 dated 19 December 1986 - and Regulamento Brasileiro da Aviação Civil (RBAC) 39, applies to all aircraft registered in the Registro Aeronáutico Brasileiro. No person may operate an aircraft to which this AD applies, unless it has previously complied with the requirements established herein.

### **AD No. 2021-02-02 - YABORÃ/EMBRAER / 39-1478.**

#### **APPLICABILITY:**

This Airworthiness Directive is applicable to Yaborã Indústria Aeronáutica S.A. / Embraer – Empresa Brasileira de Aeronáutica S.A. models EMB-145, EMB-145ER, EMB-145EU, EMB-145EP, EMB-145LR, EMB-145MR, EMB-145LU, EMB-145MP, EMB-145MK, EMB-145XR, EMB-135BJ, EMB-135ER, EMB-135KE, EMB-135KL and EMB-135LR, all serial numbers.

#### **CANCELLATION / REVISION:**

Not applicable.

#### **REASON:**

On July 2020, there was an incident involving a disconnection of a Side Arm Strut from the right Main Landing Gear in an EMB-145 model aircraft. During the subsequent investigation it was found that the MLG Side Arm Strut Lower Bearing was installed inverted in the aircraft.

Prompted by this incident, an inspection of the MLG Side Strut assembly was carried out on the operator's EMB-145 fleet. This inspection revealed more instances of incorrectly assembled (inverted) MLG Side Strut Bearings as well as instances of Bearings migration (even when the Bearings are installed correctly) and damaged/missing Bearing's sealant and potentially degraded lubrication on other aircrafts of the operator.

The inspection revealed one occurrence of an MLG Lower Side Arm installed on an aircraft possessing grease fittings on both of its sides. This is not an approved configuration as the MLG Lower Side Arm must have both of its grease fittings installed on the same side (see Figure 1 below).

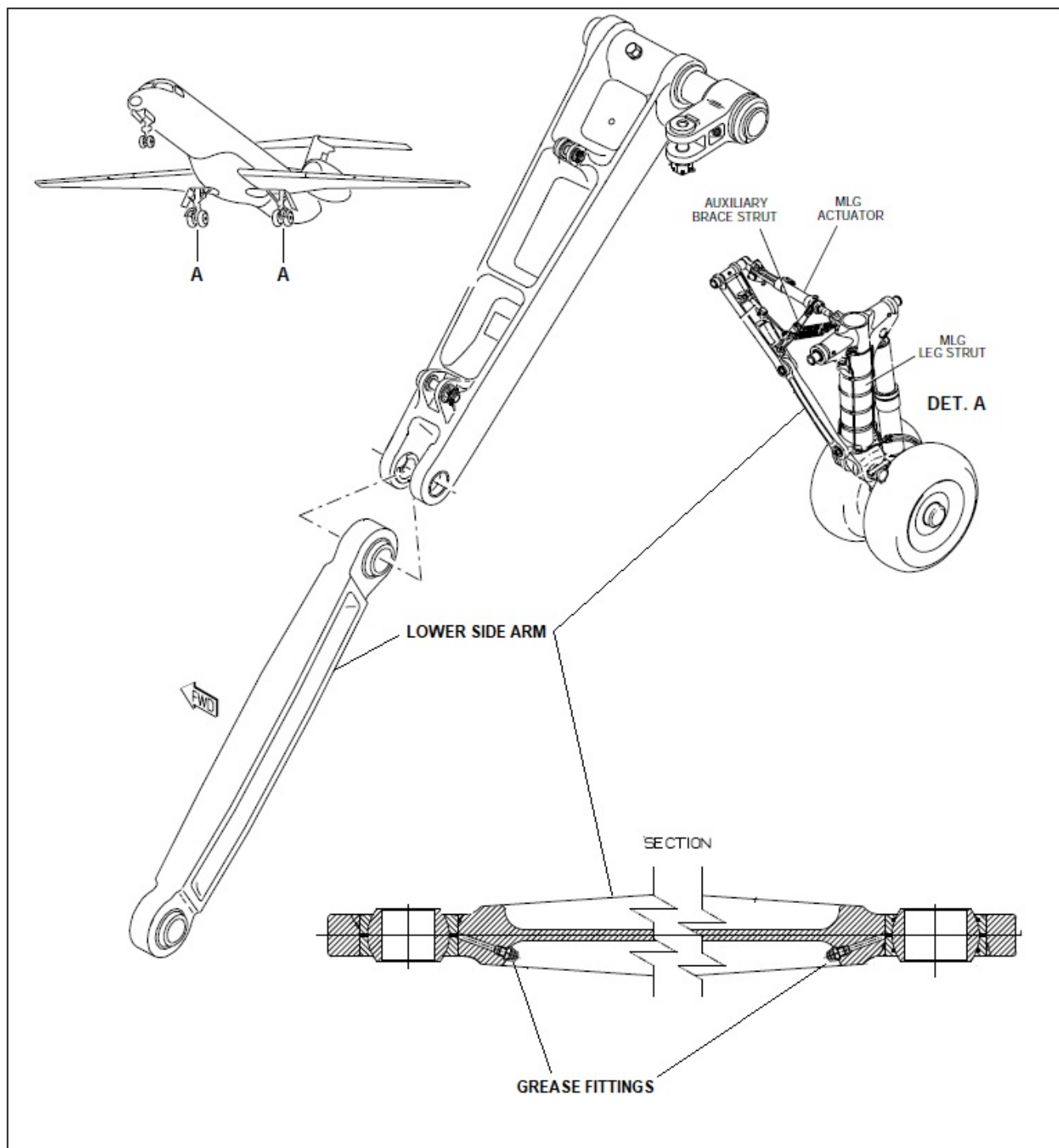


Figure 1: MLG Lower Side Arm correct grease fittings installation

An MLG Lower Side Arm with grease fittings on both of its sides could lead to a misinterpretation of the MLG assembly instructions which could culminate in an inverted installation of the MLG Side Arm Strut Bearings since the grease fittings installation side on the Lower Side Arm are used as a reference to the correct installation of the Bearings.

It was determined that the inverted installation of the MLG Side Arm Strut Bearing was a contributing factor to the disconnection of the Side Arm Strut from the Main Landing Gear.

The disconnection of the MLG Side Arm Strut could prevent the MLG to be locked in the down position, this could lead to a loss of control scenario during the landing and take-off phases. Therefore, corrective action is required, and sufficient reasons exist to mandate compliance with this AD in the indicated time limit.

#### **REQUIRED ACTION:**

For the reasons described above, this AD mandates:

A one-time inspection, and replacement of the MLG Lower Side Arm by a serviceable part (see NOTE 1), if necessary, as to the correct positioning of the grease fittings (see Figure 1).

A one-time inspection, and reinstallation of the MLG Side Arm Strut, if necessary, as to the inverted installation of the Bearings (see Figure 2).

A one-time inspection, and reinstallation of the MLG Side Arm Strut, if necessary, as to the migration of the Bearings.

NOTE 1: A serviceable MLG Lower Side Arm (P/Ns 2309-2532-001 or 2309-2532-003) is one that has both grease fittings installed on the same side of the part (see Figure 1).

NOTE 2: A Special Flight Permit (“Autorização Especial de Voo”) can be requested to ANAC (or to the Civil Aviation Authority, observing the respective regulatory code, of the state of registry of the aircraft, as applicable) to perform a ferry flight to an adequate location in order to perform the actions required by this AD.

### **COMPLIANCE:**

Required as indicated below, unless already accomplished.

#### **One-time inspections**

(a) Within the next 30 days or 150 FH (Flight Hours), after the effectivity date of this AD, whichever occurs first, visually inspect the right and left MLGs:

1. Verify that both grease fittings are installed on the same side of the MLG Lower Side Arm and that both are facing the flight direction;
  - i. If, during the inspection required by item (a)(1) above, it is found that the grease fittings are installed on opposite sides of the MLG Lower Side Arm (one grease fitting on one side and the other on the opposite side), before the next flight (see NOTE 2), replace the MLG Lower Side Arm by a serviceable part (see NOTE 1), according to the instructions of the AMM (Aircraft Maintenance Manual), procedure MPP1124\_32-10-03-04-1;
  - ii. If, during the inspection required by item (a)(1) above, it is found that both grease fittings are facing opposite to the flight direction, before the next flight (see NOTE 2), remove the Lower Side Arm and re-install it on the correct position (both grease fittings facing the flight direction – see Figure 1), according to the instructions of the AMM, procedure MPP1124\_32-10-03-04-1.
2. Verify that the flanges of the 2 MLG Side Arm Bearings are both facing the flight direction (see Details B and C in Figure 2 below);

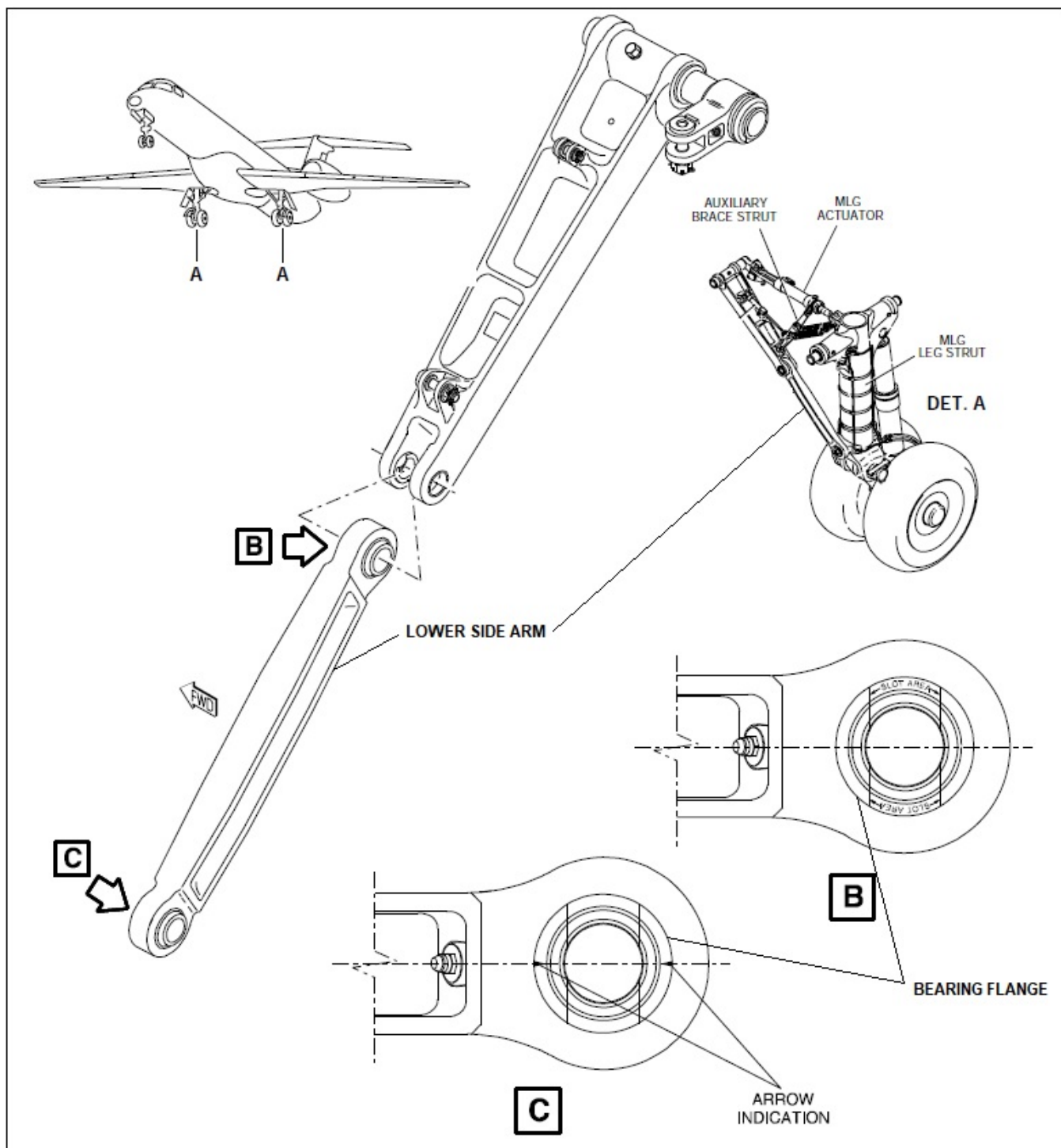


Figure 2: Correct installation of the Lower Side Arm Bearings

- i. If, during the inspection required by item (a)(2) above, it is found that either of the Bearing's flanges is installed inverted (flange facing opposite to the flight direction), before the next flight flight (see NOTE 2), remove the Lower side Arm and re-install the affected Bearing in the correct position (flange facing the flight direction), according to the instructions of the AMM, procedure MPP1124\_32-10-03-04-1.
3. Verify that there is no migration of both Bearings of the MLG Side Arm (see Details B and C in Figure 2) (both Flanges shall be flushed with the MLG Lower Side Arm face);
  - i. If, during the inspection required by item (a)(3) above, it is found that has been migration of either of the Bearings, before the next flight flight (see NOTE 2), remove the Lower Side Arm and re-install or

replace, as applicable, the affected Bearing, according to the criteria/instructions of the CMM, procedure T.P. 145/1190\_32-10-03.

### **Parts installation**

(b) After the effectivity date of this AD, do not install any MLG Lower Side Arm that is not a serviceable part (see NOTE 1).

### **Manufacturer notification**

(c) If any discrepancy is found as a result of the inspections contained in paragraph (a) of this AD, the details involved shall be communicated, within 120 hours of the finding, to:

1. ANAC – Agência Nacional de Aviação Civil through the contact information listed at the end of this AD;
2. Yaborã Indústria Aeronáutica S.A. / Embraer – Empresa Brasileira de Aeronáutica S.A.

Avenida Brigadeiro Faria Lima, nº 2170, Putim

CEP 12227-901 - São José do Campos - SP, BRAZIL.

Tel.: +55 (12) 3927-1000; E-mail:  
continued.airworthiness@embraer.net.br

### **Material Incorporated by Reference in this AD**

(d) Embraer Aircraft Maintenance Manual (AMM), Maintenance Practices and Procedures, MPP1124\_32-10-03-04-1, Revision 0, dated March 09, 2020, or later approved revisions.

(e) ELEB Component Maintenance Manual with Illustrated Parts List (CMM) – Main Side Strut, procedure T.P. 145/1190\_32-10-03, Revision 16, dated November 29, 2019, or later approved revisions.

### **Alternative methods of compliance (AMOCs)**

(f) A different method or a different compliance time to the requirements of this AD may be used if approved by the Technical Manager of the Continuing Airworthiness Technical Branch (GTAC).

### **CONTACT:**

For additional technical information, contact:

National Civil Aviation Agency (ANAC)

Continuing Airworthiness Technical Branch (GTAC)

Rua Dr. Orlando Feirabend Filho, 230 - Centro Empresarial Aquarius - Torre B - Andares 14 a 18, Parque Residencial Aquarius

CEP 12246-190– São José dos Campos - SP, BRAZIL.

Tel: +55 (12) 3203-6600; E-mail: pac@anac.gov.br

### **APPROVAL:**

Original in Portuguese language available at the Continuing Airworthiness Technical Branch (GTAC) of the National Civil Aviation Agency (ANAC). Signed by:

ROBERTO JOSÉ SILVEIRA HONORATO  
Superintendent  
Department Of Airworthiness (SAR)  
ANAC

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**Referência:** Processo nº 00066.023772/2020-15

SEI nº 5379627