



**適航指令發布單**  
**Airworthiness Directive Issuance Form**

民航局 AD 編號 AD number	CAA-2020-04-013	發布日期 Date issued	2020/4/30
適用之航空產品 Applied to (models, serial numbers or part numbers, as applicable)	Pratt & Whitney Canada (P&WC) model PW210A and PW210S engines.		
主旨摘要	Engine - Manual Calculation of Cycle Count for the Impeller and High Pressure Compressor Rotor		
民航局 CAA <input type="checkbox"/> 本國產品 Native products <input type="checkbox"/> 其他個案 Other	設計國民航主管機構 Original Authorities <input type="checkbox"/> FAA <input type="checkbox"/> EASA <input type="checkbox"/> Brazil <input checked="" type="checkbox"/> Transport Canada Civil Aviation <input type="checkbox"/> DGAC <input type="checkbox"/> Germany LBA <input type="checkbox"/> CAA-NL <input type="checkbox"/> UK CAA <input type="checkbox"/> Japan CAB <input type="checkbox"/> CAA of Israel <input type="checkbox"/> Other _____		
	設計國 AD 編號 Original AD number	CF-2020-13	
	1. 直接採用原 AD 之內容?(Is the original AD directly adopted?) <input checked="" type="checkbox"/> 是(Yes) <input type="checkbox"/> 否(No) _ a. 生效日期另訂為(Re-specify the effective date as) : _____ b. 執行時限另訂為(Re-specify the compliance time or period as) : _____ 2. 使用人是否需要將 AD 執行結果向民航局提出報告?(Do Users need to report the status of compliance to the CAA?) <input type="checkbox"/> 是(Yes) <input checked="" type="checkbox"/> 否(No)		
備註 Note	None		

註： 1. AD 內容後附。  
 2. 航空器產品使用人得向民航局提出豁免、替代符合方法、執行時限之展延之申請。  
 3. 如有任何問題，請聯絡交通部民用航空局初始適航科。Tel：(02)2349-6331~3, Fax：(02)2545-8464, e-mail：  
[adcaa@mail.caa.gov.tw](mailto:adcaa@mail.caa.gov.tw)

Note： 1. The AD text is enclosed.  
 2. Exemption, an alternative method of compliance or adjustment of the compliance time may be proposed to the CAA for approval.  
 3. For further information, please contact Civil Aeronautics Administration on Tel:(02)2349-6331~3, Fax:(02)2545-8464, e-mail：  
[adcaa@mail.caa.gov.tw](mailto:adcaa@mail.caa.gov.tw)



# AIRWORTHINESS DIRECTIVE

*This Airworthiness Directive (AD) is issued pursuant to Canadian Aviation Regulation (CAR) 521.427. No person shall conduct a take-off or permit a take-off to be conducted in an aircraft that is in their legal custody and control, unless the requirements of CAR 605.84 pertaining to ADs are met. Standard 625 - Aircraft Equipment and Maintenance Standards Appendix H provides information concerning alternative means of compliance (AMOC) with ADs.*

**Number:**

CF-2020-13

**Effective Date:**

12 May 2020

**ATA:**

72

**Type Certificate:**

E-36

**Subject:**

Engine – Manual Calculation of Cycle Count for the Impeller and High Pressure Compressor Rotor

**Applicability:**

Pratt & Whitney Canada (P&WC) model PW210A and PW210S engines.

**Compliance:**

As indicated below, unless already accomplished.

**Background:**

The engine manufacturer has discovered that the Automated Damage Tracking System (ADTS) may under-count the number of cycles accrued by the impeller and the High Pressure (HP) compressor rotor. The impeller and HP compressor rotor are both life limited components and exceeding their published life limits could result in the failure of these components.

Failure of the impeller or HP compressor rotor could result in the uncontained release of the impeller or the HP compressor rotor, and subsequently could result in damage to the engine, damage to the helicopter, and loss of control of the helicopter.

This AD mandates the use of the Manual Low Cycle Fatigue (LCF) Counting method to ensure that the impeller and HP compressor rotor do not exceed their published life limits.

This AD is considered interim action and further AD action may follow.

**Corrective Actions:**

**Part I – For engines that have accumulated 7000 starts or 14 000 flight cycles since new, whichever occurs first, as of the effective date of this AD:**

- A. Remove the impeller and HP compressor rotor upon reaching their Life Limits, as defined in Table 1 of task 00-00-00-860-801 in the applicable P&WC Engine Maintenance Manual (EMM), Part No. 30L2392 or 30L0892, Airworthiness Limitations Section, as calculated using the Manual LCF Counting method in accordance with P&WC Service Bulletins (SBs) PW210-72-A57142 or PW210-72-A57143, as applicable, both at Revision 1 and dated 26 March 2020, or later revisions approved by the Chief, Continuing Airworthiness, Transport Canada.

The use of the Manual LCF Counting method in accordance with the Original Issues of P&WC SBs PW210-72-A57142 or PW210-72-A57143, as applicable, both dated 15 January 2020, also meets the intent of paragraph A.

- B. The use of the ADTS to count the accumulated total cycles on the impeller and HP compressor rotor is prohibited. Once the accumulated total cycles of these components is established in accordance with the above paragraph A, the Manual LCF Counting method, specified in task 00-00-00-860-803 of the applicable P&WC EMM, Part No. 30L2392 or 30L0892, Airworthiness Limitations Section, must be used.

**Part II – For all engines**

Prior to removal of the engine for the purpose of sending the engine to a repair/overhaul facility, establish the accumulated total cycles of the impeller and HP compressor rotor in accordance with P&WC SBs PW210-72-A57142 or PW210-72-A57143, as applicable, both at Revision 1 and dated 26 March 2020, or later revisions approved by the Chief, Continuing Airworthiness, Transport Canada.

**Authorization:**

For the Minister of Transport,

*ORIGINAL SIGNED BY*

Rémy Knoerr  
Chief, Continuing Airworthiness  
Issued on 28 April 2020

**Contact:**

Robert Farinas, Continuing Airworthiness, Ottawa, telephone 888-663-3639, facsimile 613-996-9178 or e-mail [AD-CN@tc.gc.ca](mailto:AD-CN@tc.gc.ca) or any Transport Canada Centre.