

 適航指令發布單 Airworthiness Directive Issuance Form			
民航局 AD 編號 AD number	CAA-2017-10-008	發布日期 Date issued	2017/10/30
適用之航空產品 Applied to (models, serial numbers or part numbers, as applicable)	Airbus A318-111, A318-112, A318-121, A318-122, A319-111, A319-112, A319-113, A319-114, A319-115, A319-131, A319-132, A319-133, A320-211, A320-212, A320-214, A320-215, A320-216, A320-231, A320-232, A320-233, A320-251N, A320-271N, A321-111, A321-112, A321-131, A321-211, A321-212, A321-213, A321-231, A321-232, A321-251N, A321-253N and A321-271N aeroplanes, all manufacturer serial numbers.		
主旨摘要	Time Limits / Maintenance Checks - Safe Life Airworthiness Limitations Items - ALS Part 1 - Amendment		
民航局 CAA <input type="checkbox"/> 本國產品 Native products <input type="checkbox"/> 其他個案 Other	設計國民航主管機構 Original Authorities <div style="display: flex; justify-content: space-between;"> <div> <input type="checkbox"/> FAA <input checked="" type="checkbox"/> EASA <input type="checkbox"/> Brazil <input type="checkbox"/> Transport Canada Civil Aviation <input type="checkbox"/> DGAC </div> <div> <input type="checkbox"/> Germany LBA <input type="checkbox"/> CAA-NL <input type="checkbox"/> UK CAA <input type="checkbox"/> Japan CAB <input type="checkbox"/> CAA of Israel <input type="checkbox"/> Other _____ </div> </div>		
	設計國 AD 編號 Original AD number	2017-0215	
	1. 直接採用原 AD 之內容?(Is the original AD directly adopted?) <input checked="" type="checkbox"/> 是(Yes) <input type="checkbox"/> 否(No) _ a. 生效日期另訂為(Re-specify the effective date as) : _____ b. 執行時限另訂為(Re-specify the compliance time or period as) : _____ 2. 使用人是否需要將 AD 執行結果向民航局提出報告?(Do Users need to report the status of compliance to the CAA?) <input type="checkbox"/> 是(Yes) <input checked="" type="checkbox"/> 否(No)		
備註 Note	ATA 05. This AD supersedes EASA AD 2012-0008(CAA-2012-01-008) dated 16 January 2012 and EASA AD 2014-0141(2014-06-002) dated 04 June 2014.		
註： 1. AD 內容後附。 2. 航空器產品使用人得向民航局提出豁免、替代符合方法、執行時限之展延之申請。 3. 如有任何問題，請聯絡交通部民用航空局初始適航科。Tel：(02)2349-6331~3, Fax：(02)2545-8464, e-mail： adcaa@mail.caa.gov.tw Note： 1. The AD text is enclosed. 2. Exemption, an alternative method of compliance or adjustment of the compliance time may be proposed to the CAA for approval. 3. For further information, please contact Civil Aeronautics Administration on Tel：(02)2349-6331~3, Fax：(02)2545-8464, e-mail： adcaa@mail.caa.gov.tw			



Airworthiness Directive

AD No.: 2017-0215

Issued: 24 October 2017

Note: This Airworthiness Directive (AD) is issued by EASA, acting in accordance with Regulation (EC) 216/2008 on behalf of the European Union, its Member States and of the European third countries that participate in the activities of EASA under Article 66 of that Regulation.

This AD is issued in accordance with Regulation (EU) 748/2012, Part 21.A.3B. In accordance with Regulation (EU) 1321/2014 Annex I, Part M.A.301, the continuing airworthiness of an aircraft shall be ensured by accomplishing any applicable ADs. Consequently, no person may operate an aircraft to which an AD applies, except in accordance with the requirements of that AD, unless otherwise specified by the Agency [Regulation (EU) 1321/2014 Annex I, Part M.A.303] or agreed with the Authority of the State of Registry [Regulation (EC) 216/2008, Article 14(4) exemption].

Design Approval Holder's Name:

AIRBUS

Type/Model designation(s):

A318, A319, A320 and A321 aeroplanes

Effective Date: 07 November 2017

TCDS Number(s): EASA.A.064

Foreign AD: Not applicable

Supersedure: This AD supersedes EASA AD 2012-0008 dated 16 January 2012 and EASA AD 2014-0141 dated 04 June 2014.

ATA 05 – Time Limits / Maintenance Checks – Safe Life Airworthiness Limitations Items – ALS Part 1 – Amendment

Manufacturer(s):

Airbus (formerly Airbus Industrie)

Applicability:

Airbus A318-111, A318-112, A318-121, A318-122, A319-111, A319-112, A319-113, A319-114, A319-115, A319-131, A319-132, A319-133, A320-211, A320-212, A320-214, A320-215, A320-216, A320-231, A320-232, A320-233, A320-251N, A320-271N, A321-111, A321-112, A321-131, A321-211, A321-212, A321-213, A321-231, A321-232, A321-251N, A321-253N and A321-271N aeroplanes, all manufacturer serial numbers.

Reason:

The airworthiness limitations for Airbus A320 family aeroplanes, which are approved by EASA, are currently defined and published in the A318, A319, A320 and A321 Airworthiness Limitations Section (ALS) document(s). The Safe Life Airworthiness Limitation Items are specified in ALS Part 1. These instructions have been identified as mandatory for continued airworthiness.

Failure to accomplish these instructions could result in an unsafe condition.



Previously, EASA issued AD 2012-0008 to require the implementation of the airworthiness limitations as specified in Airbus A318/A319/A320/A321 ALS Part 1 Revision 02, and EASA AD 2014-0141 to require the implementation of specific life limits for the main landing gear (MLG) upper cardan pin Part Number (P/N) 201163620.

Since those ADs were issued, studies were conducted in the frame of in-service events or during life extension campaigns, the results of which prompted revision of the life limits of several components installed on A320 family aeroplanes. Consequently, Airbus successively issued Revision 03, Revision 04 and Revision 05 of the A318/A319/A320/A321 ALS Part 1. ALS Part 1 Revision 05 also includes the life limits required by EASA AD 2014-0141. A318/A319/A320/A321 ALS Part 1 Revision 05 issue 02 was issued to provide clarifications.

For the reason described above, this AD retains the requirements of EASA AD 2012-0008 and EASA AD 2014-0141, which are superseded, and requires accomplishment of the actions specified in A318/A319/A320/A321 ALS Part 1 Revision 05 (hereafter referred to as 'the ALS' in this AD).

Required Action(s) and Compliance Time(s):

Required as indicated, unless accomplished previously:

Replacement of Life Limited Parts:

- (1) From the effective date of this AD, replace each component before exceeding the applicable life limit, as specified in the ALS, as applicable to aeroplane model and depending on aeroplane configuration.

Aircraft Maintenance Programme (AMP) Revision:

- (2) Within 12 months after the effective date of this AD, revise the approved AMP, on the basis of which the operator or the owner ensures the continuing airworthiness of each operated aeroplane, by incorporating the limitations described in the ALS, as applicable to aeroplane model and depending on aeroplane configuration.

Credit:

- (3) For an AMP that, on the effective date of this AD, is already updated to incorporate the life limitations as specified in A318/A319/A320/A321 ALS Part 1 Revision 02, that action ensures (see Note 1 of this AD) the continued accomplishment of those limitations.

Consequently, for an aeroplane to which that AMP applies, it is acceptable to accomplish the new and more restrictive limitations, as defined in A318/A319/A320/A321 ALS Part 1 Revision 03, Revision 04 and Revision 05, respectively, to comply with paragraph (1) of this AD.

For that AMP, it is acceptable to incorporate the new and more restrictive limitations, as defined in A318/A319/A320/A321 ALS Part 1 Revision 03, Revision 04 and Revision 05, respectively, into the AMP to comply with paragraph (2) of this AD.

- (4) For an AMP that, on the effective date of this AD, is already updated to incorporate the life limitations as specified in A318/A319/A320/A321 ALS Part 1 Revision 03, that action ensures (see Note 1 of this AD) the continued accomplishment of those limitations.



Consequently, for an aeroplane to which that AMP applies, it is acceptable to accomplish the new and more restrictive limitations, as defined in A318/A319/A320/A321 ALS Part 1 Revision 04 and Revision 05, respectively, to comply with paragraph (1) of this AD.

For that AMP, it is acceptable to incorporate the new and more restrictive limitations, as defined in A318/A319/A320/A321 ALS Part 1 Revision 04 and Revision 05, respectively, into the AMP to comply with paragraph (2) of this AD.

- (5) For an AMP that, on the effective date of this AD, is already updated to incorporate the life limitations as specified in A318/A319/A320/A321 ALS Part 1 Revision 04, that action ensures (see Note 1 of this AD) the continued accomplishment of those limitations.

Consequently, for an aeroplane to which that AMP applies, it is acceptable to accomplish the new and more restrictive limitations, as defined in the ALS Part 1 Revision 05, to comply with paragraph (1) of this AD.

For that AMP, it is acceptable to incorporate the new and more restrictive limitations, as defined in the ALS Part 1 Revision 05, into the AMP to comply with paragraph (2) of this AD.

Recording AD compliance:

- (6) When the AMP of an aeroplane has been revised as required by paragraph (2), (3), (4) or (5) of this AD, as applicable, that action ensures (see Note 1 of this AD) continued accomplishment of the tasks as required by paragraph (1) of this AD for that aeroplane. Consequently, after revising the AMP, as required by paragraph (2), (3), (4) or (5) of this AD, as applicable, it is not necessary that accomplishment of individual action is recorded for demonstration of AD compliance on a continued basis.

Note 1: For affected A318, A319, A320 and A321 aeroplanes registered in Europe, complying with the approved AMP, as specified in paragraph (2), (3), (4) or (5) of this AD, as applicable, is required by Commission Regulation (EU) [1321/2014](#), Part M.A.301, paragraph 3.

Ref. Publications:

Airbus ALS Part 1 Revision 05 dated 06 April 2017, or Airbus ALS Part 1 Revision 05 issue 02 dated 19 April 2017.

The use of later approved variations or revisions of this document is acceptable for compliance with the requirements of this AD.

Remarks:

1. If requested and appropriately substantiated, EASA can approve Alternative Methods of Compliance for this AD.
2. This AD was posted on 19 September 2016 as PAD 16-134 for consultation until 17 October 2016, and republished on 26 June 2017 as PAD 16-134R1 for additional consultation until 24 July 2017. The Comment Response Document can be found in the [EASA Safety Publications Tool](#), in the compressed (zipped) file attached to the record for this AD.



3. Enquiries regarding this AD should be referred to the EASA Safety Information Section, Certification Directorate. E-mail: ADs@easa.europa.eu.
4. For any question concerning the technical content of the requirements in this AD, please contact: AIRBUS – Airworthiness Office – EIAS; Fax +33 5 61 93 44 51;
E-mail: account.airworth-eas@airbus.com.

