



適航指令發布單

Airworthiness Directive Issuance Form

民航局AD編號 AD Number	CAA-2016-08-002	發布日期 Date issued	2016/8/9
適用之航空產品 Applied to (models, serial numbers or part numbers, as applicable)	Diamond Aircraft Industries DA 42 and DA 42 M (both Normal and Restricted category) aeroplanes.		
主旨摘要	Engine Exhaust – Exhaust Pipes – Replacement		
民航局 CAA <input type="checkbox"/> 本國產品 Native products <input type="checkbox"/> 其他個案 Other	設計國民航主關機構 Original Authorities <input type="checkbox"/> FAA <input checked="" type="checkbox"/> EASA <input type="checkbox"/> Brazil <input type="checkbox"/> Transport Canada Civil Aviation <input type="checkbox"/> DGAC <input type="checkbox"/> Germany LBA <input type="checkbox"/> CAA-NL <input type="checkbox"/> UK CAA <input type="checkbox"/> Japan CAB <input type="checkbox"/> CAA of Israel <input type="checkbox"/> Other _____		
	設計國AD編號 Original AD number	2016-0156	
	1. 直接採用原AD之內容?(Is the original AD directly adopted?) <input checked="" type="checkbox"/> 是(Yes) <input type="checkbox"/> 否(No) ___ a. 生效日期另訂為(Re-specify the effective date as) : _____ b. 執行時限另訂為(Re-specify the compliance time or period as) : _____ 2. 使用人是否需要將AD執行結果向民航局提出報告?(Do Users need to report the status of compliance to the CAA?) <input type="checkbox"/> 是(Yes) <input checked="" type="checkbox"/> 否(No)		
備註 Note	DAI MSB 42-120 original issue, dated 24 June 2016. DAI WI-MSB 42-120 original issue, dated 24 June 2016. The use of later approved revisions of these documents is acceptable for compliance with the requirements of this AD.		
註：1. AD內容後附。 2. 航空器產品使用人得向民航局提出豁免、替代符合方法、執行時限之展延之申請。 3. 如有任何問題，請聯絡交通部民用航空局初始適航科。Tel：(02)2349-6331-3, Fax：(02)2545-8464, e-mail：adcaa@mail.caa.gov.tw Note：1. The AD text is enclosed. 2. Exemption, an alternative method of compliance or adjustment of the compliance time may be proposed to the CAA for approval. 3. For further information, please contact Civil Aeronautics Administration on Tel：(02)2349-6331-3, Fax：(02)2545-8464, e-mail：adcaa@mail.caa.gov.tw			



Airworthiness Directive

AD No.: 2016-0156

Issued: 02 August 2016

Note: This Airworthiness Directive (AD) is issued by EASA, acting in accordance with Regulation (EC) 216/2008 on behalf of the European Union, its Member States and of the European third countries that participate in the activities of EASA under Article 66 of that Regulation.

This AD is issued in accordance with Regulation (EU) 748/2012, Part 21.A.3B. In accordance with Regulation (EU) 1321/2014 Annex I, Part M.A.301, the continuing airworthiness of an aircraft shall be ensured by accomplishing any applicable ADs. Consequently, no person may operate an aircraft to which an AD applies, except in accordance with the requirements of that AD, unless otherwise specified by the Agency [Regulation (EU) 1321/2014 Annex I, Part M.A.303] or agreed with the Authority of the State of Registry [Regulation (EC) 216/2008, Article 14(4) exemption].

Design Approval Holder's Name:

DIAMOND AIRCRAFT INDUSTRIES GmbH

Type/Model designation(s):

DA 42 and DA 42 M aeroplanes

Effective Date: 16 August 2016

TCDS Number(s): EASA.A.005 and EASA.A.513

Foreign AD: Not applicable

Supersedure: None

ATA 78 – Engine Exhaust – Exhaust Pipes – Replacement

Manufacturer(s):

Diamond Aircraft Industries GmbH (Austria), Diamond Aircraft Industries Inc. (Canada)

Applicability:

DA 42 and DA 42 M (both Normal and Restricted category) aeroplanes, all manufacturer serial numbers, if equipped with TAE 125-02-99 engines (modification MÄM 42-198 or Optional Service Bulletin (OSB) 42-046) or TAE 125-02-114 engines (modification OÄM 42-252 or OSB 42-107).

Reason:

Two cases were reported of uncommanded engine in-flight shutdown (IFSD) on DA 42 aeroplanes. Subsequent investigations identified these occurrences were due to failure of the propeller regulating valve, caused by hot exhaust gases coming from fractured engine exhaust pipes. The initiating cracks on the exhaust pipes were not detected during previous inspections, since those exhaust pipes are equipped with non-removable heat shields that do not allow inspection for certain sections of the exhaust pipe.

This condition, if not corrected, could lead to further cases of IFSD or overheat damage, possibly resulting in a forced landing, with consequent damage to the aeroplane and injury to occupants.

To address this potential unsafe condition, Diamond Aircraft Industries (DAI) developed an exhaust pipe without a directly attached integral heat shield that allows visual inspection over the entire



exhaust pipe length. DAI issued Mandatory Service Bulletin (MSB) 42-120 and relevant Working Instruction (WI) WI-MSB 42-120, providing instructions to install the modified exhaust pipes. As an interim measure, an additional bracket was designed to hold the exhaust pipe in place in case of a pipe fracture.

For the reasons described above, this AD requires replacement of the exhaust pipes with pipes having new design, and prohibits (re)installation of the previous design pipes.

Required Action(s) and Compliance Time(s):

Required as indicated, unless accomplished previously:

Note 1: For the purpose of this AD, an “affected exhaust pipe” is an exhaust pipe having DAI P/N D60-9078-06-01, or Technify P/N 52-7810-H0001 02, or Technify P/N 52-7810-H0001 03, or Technify P/N 52-7810-H0001 04. A “not affected exhaust pipe” is an exhaust pipe having DAI P/N D60-9078-06-01_01 or Technify P/N 52-7810-H0014 01.

Note 2: For the purpose of this AD, in case the FH accumulated since first installation of an affected exhaust pipe are not known, the total time accumulated by the aeroplane since its first flight applies instead.

- (1) For an aeroplane equipped with an affected exhaust pipe that, on the effective date of this AD, does not exceed 1 300 FH since its first installation on an aeroplane, before that pipe exceeds 1 500 flight hours (FH) since its first installation on an aeroplane, replace that exhaust pipe with a not affected exhaust pipe in accordance with the instructions of Section III.1 of DAI WI-MSB 42-120 (see Note 1 and 2 of this AD).
- (2) For an aeroplane equipped with an affected exhaust pipe that, on the effective date of this AD, exceeds 1 300 FH since its first installation on an aeroplane, and does not exceed 2 800 FH since its first installation on an aeroplane, accomplish the actions as required by paragraph (2.1) or (2.2) of this AD (see Note 1 and 2 of this AD):
 - (2.1) within 200 FH or 12 months, whichever occurs first after the effective date of this AD, install additional exhaust clamps on the affected engine in accordance with the instruction of Section III.2 of DAI WI-MSB 42-120, and thereafter, before that exhaust pipe exceeds 3 000 FH since its first installation on an aeroplane, replace that exhaust pipe with a not affected exhaust pipe in accordance with the instructions of Section III.1 of DAI WI-MSB 42-120 (see Note 1 of this AD).
 - (2.2) within 200 FH or 12 months, whichever occurs first after the effective date of this AD, replace that exhaust pipe with a not affected exhaust pipe in accordance with the instructions of Section III.1 of DAI WI-MSB 42-120 (see Note 1 of this AD).
- (3) For an aeroplane equipped with an affected exhaust pipe that, on the effective date of this AD, exceeds 2 800 FH since its first installation on an aeroplane, within 200 FH or 12 months, whichever occurs first after the effective date of this AD, replace that exhaust pipe with a not affected exhaust pipe in accordance with the instructions of Section III.1 of DAI WI-MSB 42-120 (see Note 1 and 2 of this AD).



(4) Do not install on any aeroplane an affected exhaust pipe (see Note 1 of this AD) as required by paragraph (4.1) or (4.2) of this AD, as applicable.

(4.1) For an aeroplane with an affected exhaust pipe installed (see Note 1 of this AD): After modification of that aeroplane as required by paragraph (1) or (2) or (3) of this AD.

(4.2) For an aeroplane that does not have an affected exhaust pipe installed (see Note 1 of this AD): From the effective date of this AD.

Ref. Publications:

DAI MSB 42-120 original issue, dated 24 June 2016.

DAI WI-MSB 42-120 original issue, dated 24 June 2016.

The use of later approved revisions of these documents is acceptable for compliance with the requirements of this AD.

Remarks:

1. If requested and appropriately substantiated, EASA can approve Alternative Methods of Compliance for this AD.
2. This AD was posted on 01 July 2016 as PAD 16-097 for consultation until 29 July 2016. No comments were received during the consultation period.
3. Enquiries regarding this AD should be referred to the EASA Safety Information Section, Certification Directorate. E-mail: ADs@easa.europa.eu.
4. For any question concerning the technical content of the requirements in this AD, please contact: Diamond Aircraft Industries GmbH, Austria.
Telephone +43 2622 26700, Fax +43 2622 26780,
E-mail: airworthiness@diamond-air.at.

