


 <b>適航指令發布單</b> <b>Airworthiness Directive Issuance Form</b>			
民航局AD編號 AD Number	CAA-2015-05-007	發布日期 Date issued	2015/5/15
適用之航空產品 Applied to (models, serial numbers or part numbers, as applicable)	Airbus A318-111, A318-112, A318-121, A318-122, A319-111, A319-112, A319-113, A319-114, A319-115, A319-131, A319-132, A319-133, A320-211, A320-212, A320-214, A320-215, A320-216, A320-231, A320-232, A320-233, A321-111, A321-112, A321-131, A321-211, A321-212, A321-213, A321-231 and A321-232 aeroplanes, all manufacturer serial numbers.		
主旨摘要	Time Limits / Maintenance Checks – ALS Part 2 - Damage Tolerant Airworthiness Limitation Items – Implementation		
民航局 CAA  <input type="checkbox"/> 本國產品 Native products  <input type="checkbox"/> 其他個案 Other	設計國民航主關機構 Original Authorities <div style="display: flex; justify-content: space-between;"> <div> <input type="checkbox"/> FAA  <input checked="" type="checkbox"/> EASA  <input type="checkbox"/> Brazil  <input type="checkbox"/> Transport Canada Civil Aviation  <input type="checkbox"/> DGAC               </div> <div> <input type="checkbox"/> Germany LBA  <input type="checkbox"/> CAA-NL  <input type="checkbox"/> UK CAA  <input type="checkbox"/> Japan CAB  <input type="checkbox"/> CAA of Israel  <input type="checkbox"/> Other _____               </div> </div>		
	設計國AD編號 Original AD number	2015-0083	
	1. 直接採用原AD之內容?(Is the original AD directly adopted?) <input checked="" type="checkbox"/> 是(Yes) <input type="checkbox"/> 否(No) ____ a. 生效日期另訂為(Re-specify the effective date as) : _____ b. 執行時限另訂為(Re-specify the compliance time or period as) : _____ 2. 使用人是否需要將AD執行結果向民航局提出報告?(Do Users need to report the status of compliance to the CAA?) <input type="checkbox"/> 是(Yes) <input checked="" type="checkbox"/> 否(No)		
備註 Note	ATA 05. This AD supersedes EASA AD 2013-0147(CAA-2013-07-008) dated 16 July 2013.Ref. Publications:Airbus A318/A319/A320/A321 ALS Part 2 Revision 03, dated 27 October 2014. The use of later approved revisions of this document is acceptable for compliance with the requirements of this AD.		
註： 1. AD內容後附。 2. 航空器產品使用人得向民航局提出豁免、替代符合方法、執行時限之展延之申請。 3. 如有任何問題，請聯絡交通部民用航空局初始適航科。Tel：(02)2349-6331~3, Fax：(02)2545-8464, e-mail：adcaa@mail.caa.gov.tw Note： 1. The AD text is enclosed. 2. Exemption, an alternative method of compliance or adjustment of the compliance time may be proposed to the CAA for approval. 3. For further information, please contact Civil Aeronautics Administration on Tel：(02)2349-6331~3, Fax：(02)2545-8464, e-mail：adcaa@mail.caa.gov.tw			

<b>EASA</b>	<b>AIRWORTHINESS DIRECTIVE</b>
	<b>AD No.: 2015-0083</b>
	<b>Date: 12 May 2015</b>  Note: This Airworthiness Directive (AD) is issued by EASA, acting in accordance with Regulation (EC) No 216/2008 on behalf of the European Community, its Member States and of the European third countries that participate in the activities of EASA under Article 66 of that Regulation.
This AD is issued in accordance with EU 748/2012, Part 21.A.3B. In accordance with EU 1321/2014 Annex I, Part M.A.301, the continuing airworthiness of an aircraft shall be ensured by accomplishing any applicable ADs. Consequently, no person may operate an aircraft to which an AD applies, except in accordance with the requirements of that AD, unless otherwise specified by the Agency [EU 1321/2014 Annex I, Part M.A.303] or agreed with the Authority of the State of Registry [EC 216/2008, Article 14(4) exemption].	
<b>Design Approval Holder's Name:</b> AIRBUS	<b>Type/Model designation(s):</b> A318, A319, A320 and A321 aeroplanes
TCDS Number:	EASA.A.064
Foreign AD:	Not applicable
Supersedure:	This AD supersedes EASA AD 2013-0147 dated 16 July 2013.
<b>ATA 05</b>	<b>Time Limits / Maintenance Checks – ALS Part 2 - Damage Tolerant Airworthiness Limitation Items – Implementation</b>
Manufacturer(s):	Airbus (formerly Airbus Industrie)
Applicability:	Airbus A318-111, A318-112, A318-121, A318-122, A319-111, A319-112, A319-113, A319-114, A319-115, A319-131, A319-132, A319-133, A320-211, A320-212, A320-214, A320-215, A320-216, A320-231, A320-232, A320-233, A321-111, A321-112, A321-131, A321-211, A321-212, A321-213, A321-231 and A321-232 aeroplanes, all manufacturer serial numbers.
Reason:	<p>The airworthiness limitations for Airbus A320 family aeroplanes are currently included in Airbus A318/A319/A320/A321 Airworthiness Limitations Section (hereafter referred to as 'ALS') documents. The Damage Tolerant airworthiness limitation items are published in ALS Part 2, approved by EASA.</p> <p>The instructions contained in the ALS Part 2 have been identified as mandatory actions for continued airworthiness. Failure to comply with these instructions could result in an unsafe condition.</p> <p>Previously, EASA issued AD 2013-0147 to require accomplishment of all maintenance tasks as described in ALS Part 2 at Revision 02. The new ALS Part 2 Revision 03 includes new and/or more restrictive items and was approved on 27 October 2014.</p> <p>For the reason described above, this AD retains the requirements of EASA AD 2013-0147, which is superseded, and requires accomplishment of all maintenance tasks as described in ALS Part 2 at Revision 03.</p>
Effective Date:	26 May 2015

<p>Required Action(s) and Compliance Time(s):</p>	<p>Required as indicated, unless accomplished previously:</p> <p>(1) From the effective date of this AD, within the thresholds and intervals as defined in ALS Part 2 Revision 03 (see Notes 1 and 2 of this AD), accomplish all maintenance tasks, as applicable to aeroplane configuration, as specified in ALS Part 2 at Revision 03.</p> <p>Note 1: For the purpose of this AD, the thresholds and intervals as defined in ALS Part 2 Revision 3 includes the 'special' compliance times as defined in that document for certain tasks.</p> <p>Note 2: The actions specified in task 712111-01, as defined in the ALS Part 2 at Revision 03, must be accomplished as required by EASA AD 2015-0038.</p> <p>(2) If, during accomplishment of any task as required by paragraph (1) of this AD, discrepancies are detected, within the applicable compliance time as specified in ALS Part 2 Revision 03 (or in the referenced Airbus maintenance documentation), accomplish the applicable corrective action(s) in accordance with instructions specified (or incorporated by reference) in the applicable Airbus maintenance documentation.</p> <p>If no compliance time is identified in ALS Part 2 Revision 03, nor in the Airbus maintenance documentation, before next flight, accomplish the corrective action(s), or contact Airbus for approved instructions and, within the compliance time specified in those instructions, accomplish those instructions accordingly.</p> <p>If a detected discrepancy is not identified in ALS Part 2 Revision 03, nor in the Airbus maintenance documentation, before next flight, contact Airbus for approved instructions and accomplish those instructions accordingly.</p> <p>(3) Within 12 months after the effective date of this AD, revise the approved aircraft maintenance programme (AMP), on the basis of which the operator or the owner ensures the continuing airworthiness of each operated aeroplane, by incorporating the tasks and associated thresholds and intervals described in ALS Part 2 Revision 03, as applicable to the aeroplane configuration.</p> <p>(4) For an AMP that, on the effective date of this AD, is already updated to incorporate the maintenance tasks as specified in ALS Part 2 at Revision 02, the new and/or more restrictive tasks, as defined in ALS Part 2 at Revision 03, must be incorporated into the AMP to comply with paragraph (3) of this AD.</p> <p>(5) When the AMP of an aeroplane has been revised as required by paragraph (3) or (4) of this AD, as applicable, that action ensures (see Note 3 of this AD) continued accomplishment of the tasks as required by paragraphs (1) and (2) of this AD for that aircraft. Consequently, after revising the AMP, as required by paragraph (3) or (4) of this AD, as applicable, it is not necessary that accomplishment of individual tasks is recorded for demonstration of AD compliance on a continued basis.</p> <p>Note 3: For affected Airbus A318, A319, A320 and A321 aeroplanes registered in Europe, complying with the approved AMP as specified in paragraph (3) of this AD is required by Commission Regulation (EU) No <a href="#">1321/2014</a>, Part M.A.301, paragraph 3.</p>
<p>Ref. Publications:</p>	<p>Airbus A318/A319/A320/A321 ALS Part 2 Revision 03, dated 27 October 2014.</p> <p>The use of later approved revisions of this document is acceptable for compliance with the requirements of this AD.</p>
<p>Remarks:</p>	<p>1. If requested and appropriately substantiated, EASA can approve Alternative Methods of Compliance for this AD.</p>

	<ol style="list-style-type: none"><li>2. This AD was posted on 07 April 2015 as PAD 15-034 for consultation until 05 May 2015. The Comment Response Document can be found at <a href="http://ad.easa.europa.eu">http://ad.easa.europa.eu</a>.</li><li>3. Enquiries regarding this AD should be referred to the Safety Information Section, Certification Directorate, EASA. E-mail: <a href="mailto:ADs@easa.europa.eu">ADs@easa.europa.eu</a>.</li><li>4. For any question concerning the technical content of the requirements in this AD, please contact: AIRBUS – Airworthiness Office – EIAS, Fax +33 5 61 93 44 51, E-mail: <a href="mailto:account.airworth-eas@airbus.com">account.airworth-eas@airbus.com</a>.</li></ol>
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