



適航指令發布單

Airworthiness Directive Issuance Form

民航局AD編號 AD number	CAA-2024-02-007	發布日期 Date issued	2024/02/07												
適用之航空產品 Applied to (models, serial numbers or part numbers, as applicable)	Airbus A318-111, A318-112, A318-121, A318-122, A319-111, A319-112, A319-113, A319-114, A319-115, A319-131, A319-132, A319-133, A319-151N, A319-153N, A319-171N, A320-211, A320-212, A320-214, A320-215, A320-216, A320-231, A320-232, A320-233, A320-251N, A320-252N, A320-253N, A320-271N, A320-272N, A320-273N, A321-111, A321-112, A321-131, A321-211, A321-212, A321-213, A321-231, A321-232, A321-251N, A321-251NX, A321-252N, A321-252NX, A321-253N, A321-253NX, A321-271N, A321-271NX, A321-272N and A321-272NX aeroplanes, all manufacturer serial numbers.														
主旨摘要 Subject	Equipment / Furnishings - Galleys - Inspection														
民航局 CAA <input type="radio"/> 本國產品 Native product <input type="radio"/> 其他個案 Other	設計國民航主管機構 Original Authority <table><tr><td><input type="radio"/> FAA</td><td><input type="radio"/> Germany LBA</td></tr><tr><td><input checked="" type="radio"/> EASA</td><td><input type="radio"/> CAA-NL</td></tr><tr><td><input type="radio"/> Brazil</td><td><input type="radio"/> UK CAA</td></tr><tr><td><input type="radio"/> Transport Canada Civil Aviation</td><td><input type="radio"/> Japan CAB</td></tr><tr><td><input type="radio"/> DGAC</td><td><input type="radio"/> CAA of Israel</td></tr><tr><td></td><td><input type="radio"/> Other_____</td></tr></table>			<input type="radio"/> FAA	<input type="radio"/> Germany LBA	<input checked="" type="radio"/> EASA	<input type="radio"/> CAA-NL	<input type="radio"/> Brazil	<input type="radio"/> UK CAA	<input type="radio"/> Transport Canada Civil Aviation	<input type="radio"/> Japan CAB	<input type="radio"/> DGAC	<input type="radio"/> CAA of Israel		<input type="radio"/> Other_____
<input type="radio"/> FAA	<input type="radio"/> Germany LBA														
<input checked="" type="radio"/> EASA	<input type="radio"/> CAA-NL														
<input type="radio"/> Brazil	<input type="radio"/> UK CAA														
<input type="radio"/> Transport Canada Civil Aviation	<input type="radio"/> Japan CAB														
<input type="radio"/> DGAC	<input type="radio"/> CAA of Israel														
	<input type="radio"/> Other_____														
	設計國AD編號 Original AD number	2024-0038													
	1. 直接採用原AD之內容? (Is the original AD directly adopted?) <input checked="" type="radio"/> 是(Yes) <input type="radio"/> 否(No)_ a. 生效日期另訂為(Re-specify the effective date as): b. 執行時限另訂為(Re-specify the compliance time or period as): 2. 使用人是否需要將AD執行結果向民航局提出報告? (Do users need to report the status of compliance to the CAA?) <input type="radio"/> 需要(Yes) <input checked="" type="radio"/> 不需要(No)														
備註 Note	This AD supersedes EASA AD 2021-0183R1 (CAA-2021-08-001A) dated 20 September 2021.														

註： 1. AD內容後附。

2. 航空器產品使用人得向民航局提出豁免、替代符合方法、執行時限之展延之申請。

3. 如有任何問題，請聯絡交通部民用航空局初始適航科。Tel：(02)2349-6330 / 6332, Fax：(02)2545-8464,

adcaa@mail.caa.gov.tw

Note： 1. The AD text is enclosed.

2. Exemption, an alternative method of compliance or adjustment of the compliance time may be proposed to the CAA for approval.

3. For further information, please contact Civil Aeronautics Administration on Tel：(02)2349-6330 / 6332,

Fax：(02)2545-8464, adcaa@mail.caa.gov.tw



Airworthiness Directive

AD No.: 2024-0038

Issued: 05 February 2024

Note: This Airworthiness Directive (AD) is issued by EASA, acting in accordance with Regulation (EU) 2018/1139 on behalf of the European Union, its Member States and of the European third countries that participate in the activities of EASA under Article 129 of that Regulation.

This AD is issued in accordance with Regulation (EU) 748/2012, Part 21.A.3B. In accordance with Regulation (EU) 1321/2014 Annex I Part M.A.301, or Annex Vb Part ML.A.301, as applicable, the continuing airworthiness of an aircraft shall be ensured by accomplishing any applicable ADs. Consequently, no person may operate an aircraft to which an AD applies, except in accordance with the requirements of that AD, unless otherwise specified by the Agency [Regulation (EU) 1321/2014 Annex I Part M.A.303, or Annex Vb Part ML.A.303, as applicable] or agreed with the Authority of the State of Registry [Regulation (EU) 2018/1139, Article 71 exemption].

Design Approval Holder's Name:

AIRBUS S.A.S.

Type/Model designation(s):

A318, A319, A320 and A321 aeroplanes

Effective Date: 19 February 2024

TCDS Number(s): EASA.A.064

Foreign AD: Not applicable

Supersedure: This AD supersedes EASA AD 2021-0183R1 dated 20 September 2021.

ATA 25 – Equipment / Furnishings – Galleys – Inspection

Manufacturer(s):

Airbus, formerly Airbus Industrie

Applicability:

Airbus A318-111, A318-112, A318-121, A318-122, A319-111, A319-112, A319-113, A319-114, A319-115, A319-131, A319-132, A319-133, A319-151N, A319-153N, A319-171N, A320-211, A320-212, A320-214, A320-215, A320-216, A320-231, A320-232, A320-233, A320-251N, A320-252N, A320-253N, A320-271N, A320-272N, A320-273N, A321-111, A321-112, A321-131, A321-211, A321-212, A321-213, A321-231, A321-232, A321-251N, A321-251NX, A321-252N, A321-252NX, A321-253N, A321-253NX, A321-271N, A321-271NX, A321-272N and A321-272NX aeroplanes, all manufacturer serial numbers.

Definitions:

For the purpose of this AD, the following definitions apply:

Affected part: Forward-facing galleys, having a Part Number (P/N) as listed in Appendix 1 of this AD.

The inspection SB: Airbus Service Bulletin (SB) A320-25-1BVS or Airbus SB A320-25-1BVT, as applicable.



The modification SB: Airbus SB A320-25-1CBN or A320-25-1CBP, as applicable.

Aeroplane date of manufacture: The date of transfer of title (ownership) of the aeroplane at the time of first delivery to an operator, which is referenced in Airbus documentation.

Groups:

Group 1 aeroplanes are those that have an affected part installed.

Group 2 aeroplanes are those equipped with a galley which has been initially installed as an affected part on an aeroplane in production or in service in accordance with Airbus service instructions, and which following that installation, has been re-identified with a non-affected P/N without having the modification SB embodied. The identification of the galleys installed during production can be found in the Aircraft Inspection Report (AIR).

Group 3 aeroplanes are those which are not Group 2 aeroplanes and do not have an affected part installed.

Reason:

Occurrences have been reported of finding damage during inspection of certain galleys, including delamination of work deck as well as corroded and cracked retainer blocks.

This condition, if not detected and corrected, could affect the galley's capability to hold the trolley under emergency landing loads, which could lead to trolley detachment, possibly resulting in blocking of an escape path during an emergency exit.

To address this potential unsafe condition, Airbus issued the inspection SB, providing instructions to inspect the affected parts, and EASA issued AD 2021-0183 to require repetitive general visual inspections (GVI) of each affected part and, depending on findings, accomplishment of applicable corrective action(s).

After that AD was issued, it was determined that, following accomplishment of a certain repair, subsequent inspection of the repaired area could be deferred beyond the established inspection interval. Additionally, typographical errors were identified in Appendix 1 of that AD. Prompted by this development EASA issued EASA AD 2021-0183R1, accordingly.

Since EASA AD 2021-0183R1 was issued, Airbus and the galley manufacturer developed a modification to restore the design integrity of the affected galleys, and Airbus issued the modification SB, as defined in this AD, providing instructions for in-service galley modification. It was also identified that some affected galley P/N were missing in the Appendix 1 of EASA AD 2021-0183R1, while others could be removed from that list.

Since the publication of EASA PAD 23-082R1 galleys having P/N 601858-053101 and P/N 601891-011001 were added into the Appendix 1 of this AD.

For the reasons described above, this AD retains the requirements of EASA AD 2021-0183R1, which is superseded, introduces the modification SB as an optional terminating action for repetitive inspections required by this AD, and revises the list of the affected galleys.



The scope of this AD is limited to affected parts under the responsibility of Airbus as Design Organisation Approval holder. EASA considers separate mandatory actions for galleys modified or installed under the responsibility of design approval holders other than Airbus (see also Remarks section, bullet point 4 for reporting).

Required Action(s) and Compliance Time(s):

Required as indicated by this AD, unless the actions required by this AD have been already accomplished:

Inspection(s):

- (1) For Group 1 aeroplanes: Within 36 months since first installation of an affected part on an aeroplane, or within 12 months after 18 August 2021 [the effective date of the EASA AD 2021-0183 at original issue], whichever occurs later, and, thereafter, at intervals not exceeding 6 months, accomplish a GVI of each affected part in accordance with the instructions of the inspection SB (see Note 1 of this AD).

For Group 1 aeroplanes equipped with an affected part having P/N 6018A7-000101 or P/N 6018C1-000101: The initial inspection of the affected part, as required by paragraph (1) of this AD for that part, can be postponed until 12 months after the effective date of this AD.

Note 1: If the date of first installation of an affected part on an aeroplane is unknown, the aeroplane date of manufacture can be used instead.

- (2) For Group 2 aeroplanes: Within 12 months after the effective date of this AD, contact the galley manufacturer, Airbus, or the design approval holder responsible for the re-identification of the galley, as applicable, for further instructions and accomplish those instructions accordingly. The instructions must be approved by EASA or under the authority of an EASA Design Organisation Approval holder.

Corrective Action(s):

- (3) If, during any GVI as required by paragraph (1) of this AD, discrepancies are detected on an affected part, before next flight, accomplish the applicable corrective action(s) in accordance with the instructions of the inspection SB.

Part(s) Installation:

- (4) For Group 1, Group 2 and Group 3 aeroplanes: From 18 August 2021 [the effective date of the EASA AD 2021-0183 at original issue], it is allowed to install on any aeroplane an affected part, except a part having P/N 6018A7-000101 or P/N 6018C1-000101, provided that, following installation, it is inspected and, depending on findings, corrected, as required by this AD.

For Group 1, Group 2 and Group 3 aeroplanes: From the effective date of this AD, installation of an affected part having P/N 6018A7-000101 or P/N 6018C1-000101 is allowed, provided that, following installation, it is inspected and, depending on findings, corrected, as required by this AD.



GVI after Repair:

- (5) For any panel and/or retainer block of an affected part on which an intermediate repair has been accomplished in accordance with the instructions of the inspection SB, the next GVI as required by paragraph (1) of this AD can be deferred until 36 months after that repair. Thereafter, inspect that panel and/or retainer block at intervals not exceeding 6 months, as required by paragraph (1) of this AD.

Terminating Action:

- (6) Modification of an affected part on an aeroplane, accomplished in accordance with the instructions of the modification SB, constitutes terminating action for the initial and repetitive inspections as required by paragraphs (1) and (5) of this AD, as applicable, for that galley.
- (7) Modification of all affected parts on an aeroplane, accomplished in accordance with the instructions of the modification SB, constitutes terminating action for the initial and repetitive inspections as required by paragraphs (1) and (5) of this AD, as applicable, for that aeroplane, provided that no affected parts are re-installed on that aeroplane after that modification.

Acceptable Method:

- (8) For Group 1 and Group 2 aeroplanes: Replacement of an affected part with a part in post-modification SB configuration on an aeroplane is an acceptable method to comply with the requirements of paragraph (1) or (2) of this AD, as applicable, for that part.

Ref. Publications:

Airbus SB A320-25-1BVS original issue dated 18 December 2020, or Revision 1 dated 17 June 2022, or Revision 2 dated 10 August 2022.

Airbus SB A320-25-1BVT original issue dated 18 December 2020, or Revision 1 dated 17 June 2022.

Airbus SB A320-25-1CBN original issue dated 12 August 2022, or Revision 1 dated 20 March 2023.

Airbus SB A320-25-1CBP original issue dated 12 August 2022, or Revision 1 dated 20 March 2023.

The use of later approved revisions of the above-mentioned documents is acceptable for compliance with the requirements of this AD.

Remarks:

1. If requested and appropriately substantiated, EASA can approve Alternative Methods of Compliance for this AD.
2. This AD was initially posted on 06 July 2023 as PAD 23-082 for consultation until 03 August 2023 and re-published on 29 November 2023 for consultation until 27 December 2023. The Comment Response Document can be found in the [EASA Safety Publications Tool](#), in the compressed (zipped) file attached to the record for this AD.
3. Enquiries regarding this AD should be referred to the EASA Safety Information Section, Certification Directorate. E-mail: ADs@easa.europa.eu.



4. Information about any failures, malfunctions, defects or other occurrences, which may be similar to the unsafe condition addressed by this AD, and which may occur, or have occurred on a product, part or appliance not affected by this AD, can be reported to the [EU aviation safety reporting system](#). This may include reporting on the same or similar components, other than those covered by the design to which this AD applies, if the same unsafe condition can exist or may develop on an aircraft with those components installed. Such components may be installed under an FAA Parts Manufacturer Approval (PMA), Supplemental Type Certificate (STC) or other modification.
5. For any question concerning the technical content of the requirements in this AD, please contact: AIRBUS – Airworthiness Office – IIASA; E-mail: account.airworth-eas@airbus.com.



Appendix 1: Reference of the Affected Galleys

Note 2: Additional series of affected galleys, compared to EASA AD 2021-0183R1, are in **bold**. Series of affected galleys that were affected in EASA AD 2021-0183R1 but are not affected in this AD, are highlighted in ~~strikethrough~~.

Series	P/N
601537	601537-XXXXXX
601539	601539-1, 601539-101, 601539-201 and 601539-301
601557	601557-1, 601557-301
601567	601567-1, 601567-20001, 601567-201, 601567-20101, 601567-20201, 601567-20701, 601567-21201, 601567-21301, 601567-21401, 601567-401
601853	601853-000301, 601853-000401
601854	601854-XXXXXX
601855	601855-000401, 601855-000601, 601855-000701, 601855-000801, 601855-000901, 601855-001001
601856	601856-000301, 601856-000501, 601856-000601, 601856-000701, 601856-000801, 601856-000901, 601856-001001, 601856-001101, 601856-001201, 601856-001501, 601856-001601, 601856-001701, 601856-001801, 601856-001901, 601856-002001, 601856-002101, 601856-002301, 601856-002401, 601856-002501, 601856-002601, 601856-002701, 601856-002801, 601856-002901, 601856-003001, 601856-003301, 601856-003401, 601856-003501, 601856-003801, 601856-003901, 601856-004001, 601856-004101, 601856-004301, 601856-004501, 601856-004601, 601856-004701, 601856-004801, 601856-004901, 601856-005001, 601856-005101, 601856-005201, 601856-005301, 601856-005401, 601856-005501, 601856-005701, 601856-005801, 601856-005901, 601856-006001, 601856-006101, 601856-006201, 601856-006301, 601856-006401, 601856-006501, 601856-007001, 601856-007101, 601856-007501, 601856-008001, 601856-011301, 601856-011601
601857	601857-000101, 601857-000201, 601857-000301, 601857-000501, 601857-000701, 601857-000801, 601857-001001, 601857-001101, 601857-001201, 601857-001301, 601857-001501, 601857-001601, 601857-001801, 601857-001901, 601857-002001, 601857-002101, 601857-002201, 601857-002301, 601857-002501, 601857-002601, 601857-002901, 601857-003001, 601857-003101, 601857-003301, 601857-003401, 601857-003501, 601857-003601, 601857-003701, 601857-003801, 601857-003901, 601857-004001, 601857-004101, 601857-004201, 601857-004301, 601857-004401, 601857-004501, 601857-004601, 601857-004701, 601857-004801, 601857-004901, 601857-005001, 601857-005101, 601857-005201, 601857-005301, 601857-005401, 601857-005801, 601857-006001, 601857-006101, 601857-006201, 601857-006401, 601857-006501, 601857-006601, 601857-006701, 601857-006801, 601857-006901, 601857-007001, 601857-007101, 601857-007201, 601857-007301, 601857-007601, 601857-007701, 601857-007801, 601857-007901, 601857-008101, 601857-008201, 601857-008301, 601857-008401, 601857-008501, 601857-008601, 601857-008701, 601857-008801, 601857-008901, 601857-009001, 601857-009101, 601857-009201, 601857-009301, 601857-009401, 601857-009501, 601857-009601, 601857-009701, 601857-009801, 601857-009901, 601857-010001, 601857-010101, 601857-010201, 601857-010301, 601857-010401, 601857-010501, 601857-010601, 601857-010701, 601857-010801, 601857-010901, 601857-011001, 601857-011101, 601857-011201,



Series	P/N
	601857-011301, 601857-011401, 601857-011501, 601857-011601, 601857-011701, 601857-011801, 601857-011901, 601857-012001, 601857-012101, 601857-012301, 601857-012501, 601857-012601, 601857-017401, 601857-017501, 601857-017601, 601857-017701, 601857-017801, 601857-017901, 601857-018001, 601857-018101, 601857-018201, 601857-018301, 601857-018501, 601857-018601, 601857-018701, 601857-018801, 601857-019001, 601857-019101, 601857-019201, 601857-019301, 601857-019501, 601857-019601, 601857-019701, 601857-019801, 601857-019901, 601857-020101, 601857-020301, 601857-020401, 601857-020501, 601857-020701, 601857-020801, 601857-020901, 601857-021001, 601857-021101, 601857-021201, 601857-021501, 601857-021601, 601857-024701
601858	601858-000301, 601858-000401, 601858-000501, 601858-000601, 601858-000701, 601858-000801, 601858-000901, 601858-001001, 601858-001201, 601858-001301, 601858-001501, 601858-001701, 601858-001901, 601858-002001, 601858-002101, 601858-002201, 601858-002401, 601858-002501, 601858-002601, 601858-002701, 601858-002801, 601858-003001, 601858-003301, 601858-003401, 601858-003501, 601858-003601, 601858-003701, 601858-003801, 601858-003901, 601858-004101, 601858-004201, 601858-004301, 601858-004401, 601858-004601, 601858-004701, 601858-004801, 601858-004901, 601858-005001, 601858-005101, 601858-005201, 601858-005301, 601858-005401, 601858-005501, 601858-005601, 601858-005701, 601858-005801, 601858-005901, 601858-006001, 601858-006101, 601858-006201, 601858-006301, 601858-006401, 601858-006501, 601858-006601, 601858-006801, 601858-006901, 601858-007001, 601858-007101, 601858-007501, 601858-007601, 601858-007701, 601858-007801, 601858-007901, 601858-008001, 601858-008101, 601858-008201, 601858-008301, 601858-008401, 601858-008501, 601858-008601, 601858-008701, 601858-008801, 601858-008901, 601858-009001, 601858-009101, 601858-009301, 601858-009401, 601858-009501, 601858-009601, 601858-009701, 601858-009901, 601858-010001, 601858-010101, 601858-010201, 601858-010301, 601858-010401, 601858-010501, 601858-010601, 601858-010701, 601858-010801, 601858-010901, 601858-011001, 601858-011101, 601858-011201, 601858-011301, 601858-011501, 601858-011601, 601858-011701, 601858-011801, 601858-011901, 601858-012001, 601858-012101, 601858-012201, 601858-012301, 601858-012501, 601858-012601, 601858-012701, 601858-012801, 601858-012901, 601858-013101, 601858-013201, 601858-013301, 601858-013401, 601858-013501, 601858-013601, 601858-013801, 601858-013901, 601858-014101, 601858-014401, 601858-014701, 601858-014801, 601858-015001, 601858-015101, 601858-015201, 601858-015301, 601858-015401, 601858-015501, 601858-015601, 601858-015701, 601858-015801, 601858-016201, 601858-016301, 601858-016401, 601858-016501, 601858-016701, 601858-016801, 601858-016901, 601858-017001, 601858-017101, 601858-017201, 601858-017301, 601858-017401, 601858-017501, 601858-017601, 601858-017701, 601858-017801, 601858-017901, 601858-018001, 601858-018101, 601858-018201, 601858-018301, 601858-018401, 601858-018501, 601858-018601, 601858-018701, 601858-018801, 601858-018901, 601858-019001, 601858-019101, 601858-019301, 601858-019401, 601858-019501, 601858-019601, 601858-019701, 601858-019801, 601858-019901, 601858-020001, 601858-020101, 601858-020201, 601858-020301, 601858-020401, 601858-020501, 601858-020601, 601858-020701, 601858-020801, 601858-020901, 601858-021001, 601858-021101, 601858-021201, 601858-021301,



Series	P/N
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6019C1	6019C1-XXXXXX
6019C7	6019C7-000001
6019D6	6019D6-000001
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